



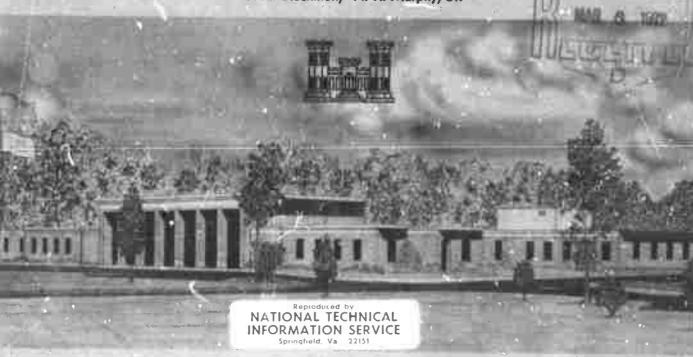
TECHNICAL REPORT NO. 3-783

AN ANALYTICAL MODEL FOR PREDICTING CROSS-COUNTRY VEHICLE PERFORMANCE

APPENDIX C: VEHICLE PERFORMANCE IN VERTICAL OBSTACLES (SURFACE GEOMETRY)

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C. A. Blackmon, N. R. Murphy, Jr.



February 1972

Spensored by Advanced Research Projects Agency and Directorate of Research,
Development and Engineering, U. S. Army Materiel Command

Service Agency U. S. Army Materiel Command

Conducted by U. S. Army Engineer Waterways Experiment Station, Vicksburg, Mississippi

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Project Nos. 1-T-0-62112-A-131 and 1-T-0-62103-A-046-02

Conducted by U. S. Army Engineer Waterways Experiment Station, Vicksburg, Mississippi

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II. SUPPLEMENTARY NOTES 2. SPONSORING MILITARY ACTIVITY Advanced Research Projects Agency and Directorate of Research, Development and Engineering, U. S. Army Materiel Command

ID. ABSTRACT

This appendix presents a brief history of vehicle dynamics modeling, a recapitulation of the U. S. Army Engineer Waterways Experiment Station approach to the problem of predicting vehicle performance in terrain containing discrete vertical obstacles, and descriptions of dynamic response prediction models for the M60Al tank and M37 truck, and compares measured and predicted vehicle performance in terms of peak vertical ned longitudinal accelerations for 78 rigid obstacle tests with the two vehicles. Ajor conclusions from the tests were that performances of the M60Al tank and M37 truck in terms of peak vertical and peak longitudinal accelerations experienced at the driver's seat when traversing discrete, rigid obstacles can be correlated with impact speed, that the mathematical techniques described yield reasonably accurate predictions of the speed at which the M60Al tank can contact a rigid obstacle without exceeding specified tolerance limits, and that refinement is needed in the dynamic response prediction model for the M37 truck.

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FOREWORD

The study reported herein was performed by the U. S. Army Engineer Waterways Experiment Station (WES) for the Office, Secretary of Defense (OSD), Advanced Research Projects Agency (ARPA), and is a portion of one task of the overall Mobility Environmental Research Study (MERS) sponsored by OSD/ARPA for which the WES was the prime contractor and the U. S. Army Materiel Command (AMC) was the service agent. broad mission of Project MERS was to determine the effects of the various features of the physical environment on the performance of crosscountry ground contact vehicles and to provide therefrom data that can be used to improve both the design and employment of such vehicles. A condition of the project was that the data be interpretable in terms of vehicle requirements for Southeast Asia. The funds employed for this study were allocated to WES through AMC under ARPA Order No. 400. funds for preparation and publication of this report were provided by the Research, Development and Engineering Directorate, AMC, under Department of the Army Project 1-T-0-62112-A-131, "Environmental Constraints on Materiel," and Project 1-T-0-62103-A-046-02, "Surface Mobility."

This appendix is one of seven to a report entitled <u>An Analytical Model</u> for <u>Predicting Cross-Country Vehicle Performance</u> (in preparation). These appendixes are:

- A. Instrumentation of Test Vehicles (July 1967)
- B. Vehicle Performance in Lateral and Longitudinal Obstacles (Vegetation)

Volume I: Lateral Obstacles (December 1968)

Volume II: Longitudinal Obstacles (July 1968)

- C. Vehicle Performance in Vertical Obstacles (Surface Geometry) (February 1972)
- D. Performance of Amphibious Vehicles in the Water-Land Interface (Hydrologic Geometry) (February 1970)
- E. Quantification of the Screening Effects of Vegetation on Driver's Vision and Vehicle Speed (April 1971)
- F. Soil-Vehicle Relations on Soft Clay Soils (Surface Composition) (August 1970)
- G. Application of Analytical Model to United States and Thailand Terrains (in preparation)

This study was conducted by personnel of the Mobility and Environmental (M&E) Division, under the general supervision of Mr. W. J. Turnbull, Technical Assistant for Soils and Engineering (retired); Mr. W. G. Shockley and Mr. S. J. Knight, Chief and Assistant Chief, respectively, M&E Division; and under the direct supervision of Mr. A. A. Rula, Chief, Vehicle Studies Branch (VSB).

The field tests reported herein were conducted during the period January-June 1968 by Mr. J. L. Gargaro, VSB, and G. Switzer, Mobility Research Branch (MRB); the mathematical models of the M60Al tank and M37 truck used to make the performance predictions were formulated and described by Mr. N. R. Murphy, Jr., MRB, and the analysis of the data presented was performed by Mr. C. A. Blackmon, VSB. This report was prepared by Messrs. Blackmon and Murphy.

Acknowledgment is made to Mr. W. A. Gross, Jr., and Mr. P. R. Gula, Development and Proof Services, Aberdeen Proving Ground (APG), for their support and assistance during the conduct of the field tank tests reported herein.

Directors of the WES during this study and the preparation of this report were COL Alex G. Sutton, Jr., CE, COL John R. Oswalt, Jr., CE, COL Levi A. Brown, CE, and COL Ernest D. Peixotto, CE. Technical Directors were Messrs. J. B. Tiffany and F. R. Brown.

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COMPUTER PROGRAM FOR SIMULATING DYNAMIC RESPONSE OF M37 TRUCK, AND DICTIONARY OF PROGRAM VARIABLES

CONVERSION FACTORS, BRITISH TO METRIC UNITS OF MEASUREMENT

British units of measurement used in this report can be converted to metric units as follows:

| Multiply | By | To Obtain |
|------------------------|------------|---------------------------------|
| inches | 2.54 | centimeters |
| feet | 0.3048 | meters |
| square inches | 6.4516 | square centimeters |
| cubic feet | 0.0283168 | cubic meters |
| pounds | 0.45359237 | kilograms |
| kips | 453.59237 | kilograms |
| short tons (2000 lb) | 907.185 | kilograms |
| pounds per square inch | 0.070307 | kilograms per square centimeter |
| foot-pounds | 0.138255 | meter-kilograms |
| miles per hour | 1.609344 | kilometers per hour |

SUMMARY

This appendix presents a brief history of vehicle dynamics modeling, a recapitulation of the U. S. Army Engineer Waterways Experiment Station approach to the problem of predicting vehicle performance in terrain containing discrete vertical obstacles, and descriptions of dynamic response prediction models for the M60Al tank and M37 truck, and compares measured and predicted vehicle performance in terms of peak vertical and longitudinal accelerations for 78 rigid obstacle tests with the two vehicles. Major conclusions from the tests were that performances of the M60Al tank and M37 truck in terms of peak vertical and peak longitudinal accelerations experienced at the driver's seat when traversing discrete, rigid obstacles can be correlated with impact speed, that the mathematical techniques described yield reasonably accurate predictions of the speed at which the M60Al tank can contact a rigid obstacle without exceeding specified tolerance limits, and that refinement is needed in the dynamic response prediction model for the M37 truck.

AN ANALYTICAL MODEL FOR PREDICTING CROSS-COUNTRY VEHICLE PERFORMANCE

APPENDIX C: VEHICLE PERFORMANCE IN VERTICAL OBSTACLES (SURFACE GEOMETRY)

PART I: INTRODUCTION

Background

- development of an analytical model for predicting the cross-country performance of a vehicle. The model was based on an energy concept within the framework of classical mechanics which requires that cause-and-effect relations be established between discrete terrain factors and vehicle response. This appendix deals with the effects of a single terrain factor--vertical obstacles. The term "obstacle" in general refers to all features of the terrain, except soil, that are inhibitory to vehicle mobility. The obstacle-effects spectrum of vehicle mobility ranges from complete immobilization to minor speed reduction. For the purpose of the overall study, obstacles were categorized according to the direction of motion forced upon a vehicle negotiating the obstacle, i.e. vertical, lateral, or longitudinal.
- 2. Perhaps the most universal single terrain feature that produces an inhibiting effect on vehicle ground mobility is small-scale surface geometry. Surface geometry features occur in a bewildering array of sizes and configurations, and produce effects on vehicles that range from "vibration" to "shock" to "immobilization," depending on the speed and size of the vehicle in relation to the size and spacing of obstacles.
- 3. Vibration-producing features are those surface irregularities of heights that can be measured in inches* rather than feet, and of

^{*} A table of factors for converting British units of measurement to metric units is presented on page ix.

distance between features that can be measured in feet rather than tens of feet. The dominant interaction for vibration-producing features is dynamic and is concentrated in the action of the vehicle suspension as the vehicle passes over the features at a speed in excess of creep speed, say 5 mph or more, with the vehicle speed limited by considerations of driver and/or cargo safety.

- 4. Shock-inducing features occur as discrete "bumps" and produce a dynamic interaction that also is concentrated in the vehicle suspension system. The dynamic action associated with these features is of a high-amplitude, low-frequency nature. The vehicle speed is ultimately limited by considerations of driver and/or cargo safety.
- 5. Features that are likely to produce immobilizations occur as discrete obstacles, as do the shock-inducing features; however, the former are of such size and shape that their negotiation can be considered in terms of static phenomena since an attempt to traverse them must be made at creep speed. The controlling factors of obstacle-vehicle interaction for these features are the geometry of the feature, the geometry of the vehicle body and running gear, and the ability of the vehicle to exert sufficient tractive effort to lift itself over the feature.
- 6. Early testing did little more than outline the problem. It was apparent that nothing less than an elaborate computer program would suffice, hence the major effort of this study was directed toward examination of current mathematical modeling techniques and the refinement thereof.

Purpose and Scope

7. This appendix presents a brief history of vehicle dynamics modeling, a discussion of the U. S. Army Engineer Waterways Experiment Station (WES) approach to the problem of predicting vehicle performance when traversing discrete vertical obstacles, a description of the dynamics submodels used, and a comparison of measured and predicted vehicle performance.

8. Seventy-eight obstacle-vehicle tests were run with two vehicles at several speeds over a range of obstacle heights. Dynamic response predictions in terms of peak vertical acceleration and peak longitudinal acceleration were made for 34 combinations of obstacle height and vehicle speed. The tests and predictions were limited to the vehicles crossing a rigid, nondeformable obstacle on a uniformly hard surface.

A History of Vehicle Dynamics Modeling

In retrospect

- 9. Since the advent of the automobile (especially since the 1920's), research by government and industry has been conducted on a continuing basis in various countries with regard to highway design and construction and also vehicle design, with particular emphasis on steering control, power train, and suspension -- the principal contributors to the safety, efficiency, and riding comfort of on-road vehicles.
- 10. The speed at which a driver of a vehicle will traverse obstacles or continuous irregular terrain is controlled primarily by the level of vibration activity that does not exceed his particular ride comfort level. Vehicle vibration or ride is sensed by a driver or passenger through sight, touch, and hearing in response to external stimuli, such as motions, forces, and sounds. Whenever this sensation becomes too severe, the driver will alter the vehicle's speed until the sensation reaches an acceptable level. This sensation, therefore, is a significant factor in determining the speed of a vehicle over a given terrain. The irregular terrain-vehicle problem is essentially one of dynamics, and its solution must include the combined effects of the surface being traversed, the vehicle, and the driver. Because of the complexity of the problem and the desire to produce better riding vehicles, considerable effort has been expended on modeling dynamic vehicle response.
 - 11. Because of the lack of mathematical techniques required in modeling suspension systems, much of the early work consisted of

cut-and-try methods. The first significant contributions to an analytical treatment of vehicle dynamics were performed by Rowell, 1 Guest, 2 and Olley 3 in the early 1920's and 30's.

- Schilling and Fuchs specifically for suspension analysis, and although it was suited to only a single-degree-of-freedom system, it did permit the inclusion of the nonlinear characteristic of shock absorbers. The analyzer was used in the continuous determination of transient motions and in the portrayal of the effect on motion by changes in the characteristics of the shock absorber. This differential analyzer was the forerunner of today's analog computer, and its development led to rapid advances in suspension analysis and design. By the 1950's, it was widely exploited by the automotive industry.
- 13. In 1953, Jeska developed a four-degree-of-freedom model that included pitch and bounce of the body and vertical motions of the front and rear wheels. The forcing function was an actual road wave measured by a photographic technique. In 1955, Bodeau, Bollinger, and Lipkin of Ford Motor Company developed a detailed ride analysis in which a nine-degree-of-freedom model was used to describe a passenger car. In 1960, Kohr of General Motors Corporation developed a mathematical simulation of automobile ride. In his simulation, a measured road profile was recorded on magnetic tape, and the tape was fed through an analog computer model of the vehicle to predict the vehicle motions, i.e. pitch, bounce, and roll. The resulting motions were used to drive a vibration simulator, which was used as a laboratory means of assessing the effect of the vibration on humans.
- 14. Until about 1960, the analysis of ride had been concerned primarily with the suspension system and means of improving the ride quality. Although considerable work was done in the area of human tolerance to vibration, a means for quantifying human tolerance to vibrations had not been developed. Van Deusen has shown that very little of the research done actually pertains to the off-road environment. Most experiments have been devised to assess human response to sinusoidal motion in only one direction, while the more complex ride comfort

problems involve random vibrations in various directions. The most frequently used criteria have been those of Dieckman and Janeway, who developed simple formulas for relating comfort limits to amplitude and frequency of vibration. There have been several studies of "onthe-road" measurements of ride comfort. For example, Von Eldik Thiemell examined the Dieckman-Janeway criteria in the actual vehicle environment, but he met with little success. Van Deusen 22,13 and Versace used a technique, referred to as cross modality, in which subjects received noise signals through earphones and adjusted and matched the signal's level to the sensation level of ride vibration. A statistical analysis showed favorable correlations of the measured accelerations with ride sensation, and at least indicated that correlations between ride sensation and vibration were possible.

- 15. In the late 1950's the Department of Defense began to recognize the significance of vehicle vibration on off-road mobility and weapon efficiency. An extensive effort was begun to quantify the vehicle vibration problem and to correlate it with human response and terrain characteristics. Interest was shifted from deterministic to stochastic techniques. The latter technique consists of classifying terrain profiles by certain pertinent statistics and analyzing the response statistically. The groundwork for this type of analysis was begun in 1959 by Bogdanoff and Kozin, 15 who described in detail the statistical analysis of the responses of simple linear systems to random terrain inputs. Although the vehicle models were simple and idealized, the analyses provided a starting point and yielded much useful information regarding fundamental relations between pertinent vehicle parameters and statistical terrain quantities. This study preceded studies of Bieniek 16 (1960), Van Deusen 17 (1962), and Bussman 18 (1964), who followed essentially the same approach as that described by Bogdanoff and Kozin. The one notable exception was Van Deusen's introduction of a nonlinear vehicle system into his statistical analysis.
- 16. Organized discrete obstacle-vehicle research in the Western world was given special attention as the result of World War II experiences. Early U. S. military efforts were concerned with designing

military vehicles that would reduce immobilizations caused by obstacle interference. Obstacle test courses were constructed at Aberdeen Proving Ground (APG), and tests on these courses have become part of the overall vehicle engineering evaluation test program. Results of these studies led to the recent development of articulated vehicles. Discrete obstacle-vehicle research studies in the United States gained more emphasis about the mid-1950's when terrain factors other than soils were introduced as deterrents to off-road vehicle travel, and more attention was given to obstacle geometry interference and the effects of dynamic response on vehicle performance. By the early 1960's these studies produced several static and quasi-dynamic models which related, by simple two-dimensional static mechanics, slope and obstacle geometry to go-no go performance.

- 17. In 1963, the U. S. Army Tank-Automotive Command (TACOM) began basic research on the effects of vehicle vibration on human response. This work was based on the results of past studies and so was oriented toward quantifying the effects of random vibration on vehicle-driver performance. Two performance parameters were developed to describe human response--acceleration density and absorbed power. Of the two performance parameters, absorbed power is preferred in quantifying human response to vibration since it is a descriptor of the flow of energy from the vibrating vehicle to the driver. This led to the TACOM V-ride concept in which ride limiting speed is determined as that speed at which the driver's absorbed power reaches 6 watts. During the 1960's the Department of Defense sponsored several studies in the development and application of ad hoc comprehensive cross-country models in which V-ride was incorporated as a submodel.
- 18. In the early 1960's the scope of WES mobility research was expanded, and static and dynamic surface configuration, vehicle, driver interaction studies were initiated. In 1965, FMC Corporation conducted a study ¹⁹ for WES to determine the feasibility of using a digital computer to simulate the dynamic response of ground vehicles traveling over unvielding irregular terrain segments. This study resulted in the development of a generalized mathematical model of an n-axle vehicle

which, within limits, is suitable for both wheeled and tracked vehicles. Current approaches

- Today's practice in modeling the effects of surface config-19. uration on vehicle performance consists essentially of two types of analysis which are separated on the basis of the kind of vehicle and/or driver interaction anticipated. Regardless of the analysis performed, the terrain profile and associated discrete obstacles generally are considered rigid. This consideration represents the worst conditions that might be encountered from the standpoint of the vehicle's vibrational behavior. If a terrain unit contains an irregular surface that can be easily overridden by a vehicle without inducing frequent shock, vehicle performance is predicted by a dynamic model. In this case, the problem is commonly identified as surface roughness, and the profile used is a statistically uniform surface profile. If a terrain unit contains discrete obstacles larger than those included in the rough terrain analysis and which are likely to produce immobilizations, it is assumed that traversing or circumventing the obstacles will be accomplished at a creep speed and the interactions are treated as static phenomena. The models used in such terrain situations are thus static models. The controlling factors in the relation of a static model are the geometry of the obstacle and vehicle configuration. If the discrete obstacles that the vehicle must pass over occur at wide spacings (e.g. dikes), and they can be overridden at speeds greater than creep speeds, a dynamic model is used to determine the maximum override speed.
- 20. Current vehicle dynamic models simulate mathematically the dynamic response of selected points within a vehicle (usually at the driver's seat or in the cargo compartment) as it traverses discrete obstacles or rough terrain. Performance is generally expressed in terms of relations between speed and such response quantities as absorbed power, root-mean-square acceleration, or peak acceleration, and then referenced to established tolerances for horizontal and vertical accelerations and power limits. Most dynamic models predict only vertical motions; however, they can readily be modified to include horizontal and, if necessary, lateral motions. The mathematical techniques

involved in the formulation of dynamic models are common to all models, but differences occur in the details of representing the terrain, vehicle, and driver limits, and in the size of the computer required.

21. Mathematical descriptions of vehicle behavior in surmounting obstacles are a combination of dynamic and static models. Discrete obstacles such as rocks, boulders, mounds, scarps, ditches, etc., are usually first examined to determine whether or not there will be spatial interference between the obstacles and the nonpropelling vehicle structure. This examination may proceed in either two or three dimensions: with or without compliance of the vehicle running gear, suspension, or structure; with or without compliance of the obstacle itself; or by spatial matching of the vehicle and the obstacle, either of which may be described more or less completely. In some instances, the vehicle underside is modeled to scale, usually in two dimensions. In others, the vehicle shape is idealized to the quantitative description and location of salient features, and matching is done through the application of complex but ordinary trigonometry and geometry. The latter procedure is, of course, more suitable for computer use. Where the number of obstacle configurations assumed in an area is relatively small, however, the scale-model experiments can be conducted once and for all, and the results stored for subsequent computer consultation as needed. In addition to static examination regarding obstacle-vehicle geometry interference aspects, it is necessary also to describe the dynamic response of the vehicle when traversing a given obstacle at a given speed. It is this problem upon which this report is focused.

PART II: THE WES APPROACH

Obstacle-Vehicle Interaction Categories

22. Early efforts at WES were directed toward developing instrumentation 20 and test procedures to measure and record vehicle response. It was apparent that a model for predicting the effects of vertical obstacles must consider at least three broad categories of obstaclevehicle interactions.

Obstacle geometry interference

23. This model was designed to answer such questions as: Is it possible for the vehicle to cross the obstacle without hanging up? Is the obstacle of such size and configuration that the vehicle might be in danger of up-ending? Is there sufficient traction? Initially, a two-dimensional scale model of the vehicle and of the obstacle was used to answer the first two questions and to determine a maximum attitude angle which was compared with the maximum negotiable slope to determine a simple go-no go. A procedure for determining maximum slope negotiable is given in Appendix D²¹ of this report. Subsequently, a computer program was written that mathematically determined the same performance parameters.

Speed limited by maneuvering

24. A method for determining whether or not the vehicle can circumvent the obstacle on the basis of a theoretical parameter, "area denied," and empirical relations of area denied and speed made good were developed to answer such questions as: Is it possible for the vehicle to circumvent the obstacles? What reduction in speed is brought about if the vehicle does circumvent the obstacles? This method is described in Appendix B, Volume I, 22 of this report. A computer program was written that utilized a somewhat refined procedure for computing area denied and the effect of area denied on vehicle speed.

Speed limited by dynamic response

25. If the vehicle can traverse the obstacle, the maximum safe

speed at which it can cross the obstacle must be determined. this, the performance parameters and conditions that limit speed must be known. A system for predicting vehicle dynamic response was being developed concurrently with the conduct of the MERS vehicle field test programs, but the input and output requirements of the prediction system were not known at the time the test programs were conducted. Consequently, the response data taken during the MERS test programs did not entirely satisfy the requirements of the prediction system to the extent that they could be used for comparison with predicted values. However, the test results did tend to confirm previous cross-country studies. 23 which show that the maximum peak vertical acceleration tolerated by the driver while trying to maintain a maximum safe speed was approximately 2.5 g's. An additional tolerance limit of 2.0-g peak longitudinal acceleration was established on the basis of competent experience and judgment. Other tests²⁴ confirmed that peak vertical acceleration and peak longitudinal acceleration could be adequately measured in vehicle tests. Computer models for mathematically simulating vehicle dynamic response were developed to predict peak vertical acceleration and peak longitudinal acceleration for rigid-frame wheeled and tracked vehicles crossing discrete obstacles. Descriptions of the models are discussed in subsequent paragraphs.

Dynamic Response Prediction Model for M60Al Tank

- 26. A mathematical model of the M60Al tank was developed to simulate the dynamic response of the tank while crossing rigid obstacles perpendicular to the path of travel. The intent of this model was to portray as nearly as possible the significant features of the motions of interest, namely, those in the vicinity of the driver, and to predict obstacle limiting speeds based on some preselected tolerance criterion at a minimum cost. The model thus does not contain the detail that might be required, for example, in designing a vehicle suspension system.
 - 27. A schematic diagram of the system that was modeled is shown

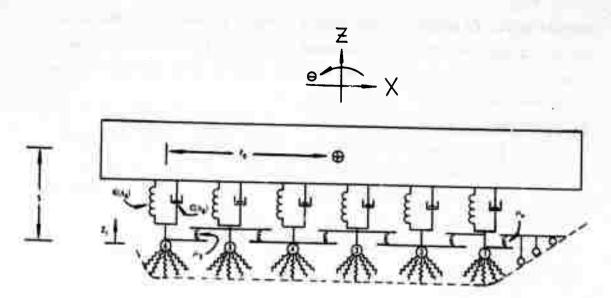


Fig. Cl. Schematic of M60Al tank

in fig. Cl. This model consists of nine degrees of freedom that include the bounce, pitch, and surge of the center of gravity of the main frame and the vertical motions of each of the six bogie wheels. Motion in this context includes displacement, velocity, and acceleration. In addition, the motions in the vicinity of the driver are computed.* The geometry effects of the bogies are represented by radially projecting stiff springs and the track compliance by interconnecting springs between the bogies and three "feelers" appropriately positioned in front of the first bogie and connected to it by a spring.

28. The longitudinal motion (X in reference axis, fig. Cl) is accounted for only in the acceleration determined from the horizontal forces resulting from deflections of the bogie spring segments. The horizontal components of the segment forces are summed for each bogie, and the horizontal acceleration is obtained by dividing this summation by the mass of the tank. This method of accounting for horizontal

^{*} Because of certain geometric interferences, it is not possible to locate an accelerometer at the driver position. An accelerometer intended to measure the accelerations in the vicinity of the driver can be conveniently located at a position 1 ft to the right and 2 ft behind the center of the driver's seat.

accelerations is somewhat analogous to towing a vehicle across an obstacle, always maintaining a constant velocity, and determining the increased towing force required to tow the vehicle over the obstacle at the given velocity.

Equations of motion

29. The differential equations describing the motion of this system were developed by first establishing an appropriate set of coordinates and sign convention and then placing the system in a displaced configuration such that each coordinate was affected. The relative displacements of the masses produce forces on each mass as shown by the free body diagram in fig. C2. The vehicle characteristics used in

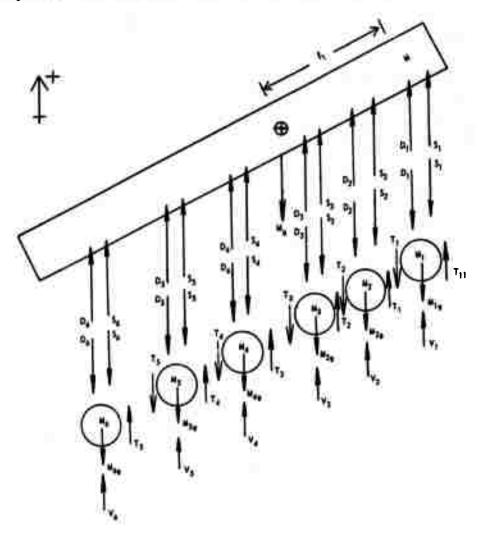


Fig. C2. Vertical forces acting on M60Al tank free body

predicting vehicle dynamic response are given in table Cl. Using Newton's second law of motion and summing the forces on the main frame and on each bogie led to the series of equations listed below.

Forces on the main frame (sprung mass):

$$M\ddot{z} = -\left[\sum_{i=1}^{6} k(\Delta_{i})\Delta_{i} + \sum_{i=1}^{6} c(\Delta_{i})\Delta_{i} + Mg\right]$$

$$\vdots = -\left[\sum_{i=1}^{3} k(\Delta_{i})\Delta_{i}\ell_{i} \cos \theta + \sum_{i=1}^{3} c(\Delta_{i})\Delta_{i}\ell_{i} \cos \theta - \sum_{i=3}^{6} k(\Delta_{i})\Delta_{i}\ell_{i} \cos \theta - \sum_{i=3}^{6} c(\Delta_{i})\Delta_{i}\ell_{i} \cos \theta + Q\right]$$

$$M\ddot{x} = \sum_{i=1}^{6} H_{i}$$

Vertical forces on the bogies (unsprung mass):

$$\begin{split} &M_{1}\ddot{z}_{1} = k(\Delta_{1})\Delta_{1} + c(\dot{\Delta}_{1})\dot{\Delta}_{1} + \mu_{1}\delta_{1} - \mu_{0}\delta_{0} - M_{1}g + V_{1} \\ &M_{2}\ddot{z}_{2} = k(\Delta_{2})\Delta_{2} + c(\dot{\Delta}_{2})\dot{\Delta}_{2} - \mu_{1}\delta_{1} + \mu_{2}\delta_{2} - M_{2}g + V_{2} \\ &M_{3}\ddot{z}_{3} = k(\Delta_{3})\Delta_{3} + c(\dot{\Delta}_{3})\dot{\Delta}_{3} - \mu_{2}\delta_{2} + \mu_{3}\delta_{3} - M_{3}g + V_{3} \\ &M_{4}\ddot{z}_{4} = k(\Delta_{4})\Delta_{4} + c(\dot{\Delta}_{4})\dot{\Delta}_{4} - \mu_{3}\delta_{3} + \mu_{4}\delta_{4} - M_{4}g + V_{4} \\ &M_{5}\ddot{z}_{5} = k(\Delta_{5})\Delta_{5} + c(\dot{\Delta}_{5})\dot{\Delta}_{5} - \mu_{4}\delta_{4} + \mu_{5}\delta_{5} - M_{5}g + V_{5} \\ &M_{6}\ddot{z}_{6} = k(\Delta_{6})\Delta_{6} + c(\dot{\Delta}_{6})\dot{\Delta}_{6} - \mu_{5}\delta_{5} - M_{6}g + V_{6} \end{split}$$

where

for
$$1 \le i \le 3$$
 $\triangle_i = z + \ell_i \sin \theta - z_i$, $\triangle_i = z + \ell_i \theta \cos \theta - z_i$
 $4 \le i \le 6$ $\triangle_i = z - \ell_i \sin \theta - z_i$, $\triangle_i = z - \ell_i \theta \cos \theta - z_i$

and

Q = moment about the center of gravity of the main frame produced by horizontal forces, Q = $\sum_{i=1}^{6} (H_i)(S + \Delta_i)$

l = distance from center of gravity of main frame to contact
point of ith bogie

 $k(\triangle_{i})$ = force-deflection relation for ith bogie suspension (fig. C3)

 $c(\triangle_i)$ = force-velocity relation for ith bogie suspension (fig. C4)

V_i = resultant vertical force of spring segments of ith bogie

H_i = resultant horizontal force of spring segments of ith
 bogie

 $\delta_{i} = z_{i+1} - z_{i} = relative displacement between adjacent bogies$

 μ_i = spring constant for ith track spring; in this study, for $1 \le i \le 5$ μ_i = 375 lb/in., μ_0 = 600 lb/in.

30. Observation of photographs of the tank crossing the highest obstacle (18 in.) revealed that the greatest pitch angle expected would be on the order of 9 deg or less. It is seen that if

$$\theta = 9 \deg$$

then

$$\cos \theta = \cos 9 \deg = 0.988 \approx 1$$

and

9 deg =
$$\frac{\pi}{20}$$
 = 0.157 radian

Since

$$\sin 9 \deg = 0.156 \approx 0.157$$

the small angle assumption, i.e. $\cos \theta = 1$, $\sin \theta = \theta$, is valid. For this reason, and to simplify the calculations, the small angle concept was used in the equations above.

31. Once the motions at the center of gravity of the main frame have been determined, the motions in the vicinity of the driver can be

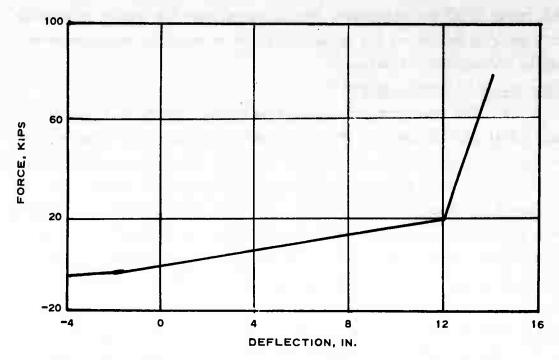


Fig. C3. M60Al suspension spring force versus deflection

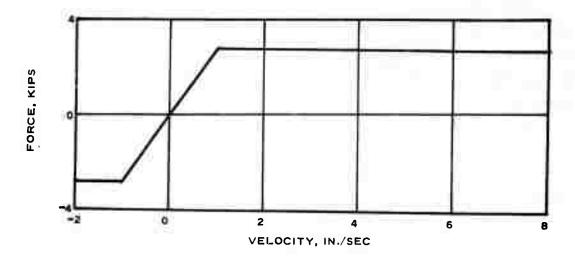


Fig. C4. M60Al suspension damping force versus velocity

determined by combining the translatory and rotational motions to yield the equation:

$$Z_{DR} = Z_{CG} + 35\theta$$

The value of 35 in. represents the distance from the center of gravity to a point 2 ft behind the driver, a point at which an accelerometer can be conveniently located.

Computation of track forces

- The track compliance is represented chiefly by interconnecting linear springs between the bogies and three "feelers" that are connected to the front bogie by a stiff spring. The spring constants used in this study were determined by observing photographs of the tank in different positions on an obstacle. From these photographs, estimates were made of the influence on the displacement of adjacent bogies of displacing a particular bogie. Knowing the approximate mass of each bogie assembly, an appropriate spring constart could be determined. Close observation further revealed that upon approaching an obstacle larger than about 6 in. high the initial track-obstacle contact tended to lift the front bogie and guide it over the obstacle. This lifting has a significant effect on the longitudinal motion. To simulate this effect, three feelers were positioned in front of the first bogie, each at a different threshold height, to conform with the geometry of the leading portion of the track. The influence of the feelers in lifting the front bogie depends on the height and shape of the encountering obstacle. Since no information was available to enable the determination of an effective spring constant, an arbitrary value of 600 lb/in. was chosen. Although this value was estimated, it shows that with a proper determination of a spring constant these longitudinal accelerations can be adequately simulated.
- Bogie spring segments
- 33. The segmented wheel concept 25 was used in the model to (a) enable predictions of longitudinal accelerations, (b) include important geometry effects of the bogies, and (c) incorporate a means for describing the composite compliance of the real bogie-track-obstacle sys-This composite compliance of the real system includes such phenomena as the effects of the small terrain and obstacle deformations, track deformations, and envelopment characteristics among others that otherwise in the model would be represented as infinitely rigid.

34. Each bogie was divided into twelve 10-deg segments, six on each side of the vertical position as shown in fig. C5. To account for

DIRECTION OF TRAVEL

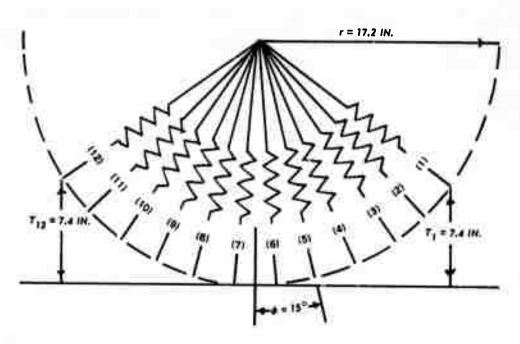


Fig. C5. Schematic showing bogie spring segment configuration

the track thickness, an effective bogie radius of 17.2 in. was used. At these conditions, an average horizontal spacing of 2.357 in. coincided closely with each spring position.

Determination of threshold heights

35. To compute the deflections of each spring segment, the segment threshold heights, $T_{\bf i}$, were first computed. These heights are simply the heights to each spring of the undeflected wheel (see fig. C5). The segment deflection, $\xi_{\bf i}$, is then computed by the equation:

$$\xi_{i} = \begin{cases} Y_{i} - T_{i} - z_{i}, Y_{i} - T_{i} - z_{i} \geq 0 \\ 0, Y_{i} - T_{i} - z_{i} < 0 \end{cases}$$

where

Y_i = vertical obstacle height beneath the ith segment z_i = vertical axle displacement of ith bogie

36. The segment deflections are permitted to have positive values only; negative values are replaced by zero. The reference from which vertical displacements are measured is the point that locates the bogic axle when the bogic is imagined to be rigid and in static equilibrium. Static deviations from this reference correspond to static wheel deflections, and superposed on these static deflections are the dynamic obstacle-induced deflections.

Computation of vertical and horizontal forces

37. The resultant vertical and horizontal forces on the first bogie axle due to the spring segment deflections are given by equations:

$$V_{1} = \sum_{i=1}^{12} (k_{v} \cos \emptyset_{i}) \xi_{i}$$

$$H_{1} = \sum_{j=1}^{12} (k_{h} \sin \emptyset_{i}) \xi_{i}$$

where

k_v and k_h = segment spring constants for the vertical and horizontal modes, respectively

Ø_i = angle of the ith segment from the vertical
ξ_i = vertical deflection of the ith segment

38. Generally, k_v and k_h would have the same value. However, the increased stiffness noted in the horizontal mode warranted a higher spring constant in the horizontal mode. These values, k_v and k_h, were determined by examining oscillograph records for several tests over a 10-in.-high obstacle. These k values were adjusted until the model outputs adequately portrayed the gross features of the acceleration-time histories from the oscillographs. The choice of the 10-in. obstacle was quite arbitrary; it was chosen solely because it was close to the median obstacle height that the M60Al tank would be expected to traverse. Ideally, such k values would be determined by

appropriately instrumenting the bogies and performing a series of systematic tests that would lead to the determination of a proper set of values.

39. Defining $\gamma_i = k_v \cos \theta_i$ and $\sigma_i = k_h \sin \theta_i$, a γ -array and a σ -array are established in the same manner as the array of threshold heights, thus simplifying the computations of the resultant vertical and horizontal forces.

Computation of moment produced by horizontal forces

40. An important contribution to the moments about the center of gravity of the main frame is the horizontal forces acting on the bogies. A schematic diagram, showing only the front and rear suspensions, is given in fig. C6 to illustrate the manner in which the horizontal forces contribute to the moment. Using the small angle assumption and the established sign convention that the suspension deflection, Δ , is negative in the equilibrium position, the following equation is used to compute the moment, Q, due to horizontal forces.

$$Q = -\sum_{i=1}^{3} (H_{i}) [(S + \Delta_{i}) - (\ell_{i})\theta] - \sum_{i=4}^{6} (H_{i}) [(S + \Delta_{i}) + (\ell_{i})\theta]$$

where

the quantities in brackets represent the moment arm from the center of gravity to the ith bogie

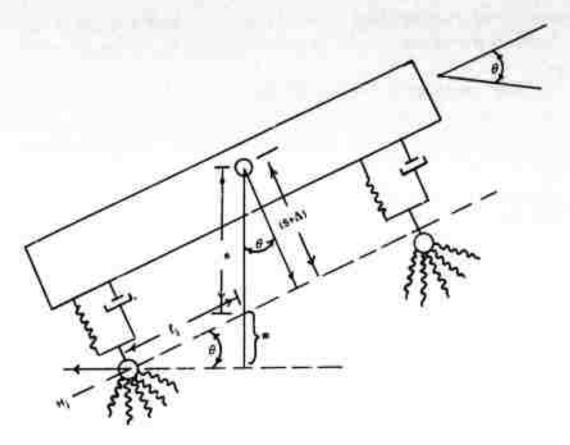
S = vertical distance from the center of gravity to the bogie axles in the undeflected state

H_i = resultant horizontal force of the ith bogie

l_i = longitudinal distance from the center of gravity to the
 ith bogie

Computer program

41. A complete listing of the digital computer program and a dictionary of the computer variables used are given at the end of this appendix. This is a FOETRAN IV program written for a GE-430 time-sharing system. Mixed mode operations, which are acceptable in this system, were used on occasion where it proved advantageous in reducing



MOMENT AT CENTER OF GRAVITY OF FORCE, $H_i = H_i (m + n)$ WHERE

$$n = \frac{s + \Delta}{\cos \theta} = s + \Delta$$

$$m = \ell_i SIN \theta = \ell_i \theta$$

$$Q_i = H_i[(S + \Delta_i) + \ell_i \theta]$$

Fig. C6. Schematic showing moment of horizontal forces

the logic or the number of required statements. The file entitled "Fimake" serves as a convenient method to build and input the obstacles.

Dynamic Response Prediction Model for M37 Truck

42. The development of the mathematical model of the M37 truck to simulate the dynamic response of the truck while crossing rigid obstacles followed the same reasoning as that given in paragraph 26 for the M60Al tank.

43. A schematic diagram of the system that was modeled is given in fig. C7. This model consists of four degrees of freedom that

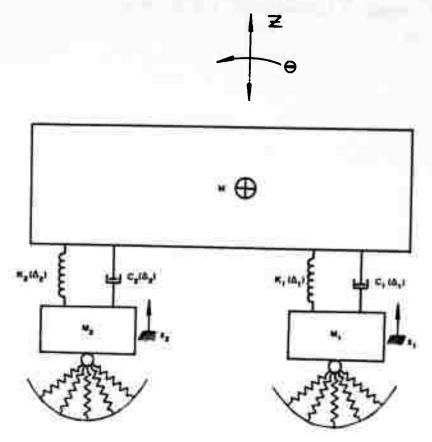


Fig. C7. Schematic of truck model

include bounce and pitch of the center of gravity of the main frame and the vertical motions of both axles. A significant difference in the two mathematical models is that longitudinal motion is not taken into account in the M37 truck model. The motions in the vicinity of the driver* are computed from the predicted motions at the center of gravity. The frame of the truck was considered rigid, and only the pneumatic tires and suspensions were considered to contribute to the sprung motion of the frame. The model includes all pertinent nonlinearities in the suspension. Vehicle characteristics used are given in table C2.

^{*} Accelerometers can be conveniently located underneath the center of the driver's seat.

Equations of motion

44. The differential equations describing the motions of this system were developed in a manner analogous to that described in paragraph 29. Again, using Newton's second law of motion and summing the forces on the body and on each axle led to the series of equations listed below.

Forces on body (sprung mass):

$$\begin{split} \text{Mz} &= \text{K}_{1}(\triangle_{1})(z_{1} - z - a \sin \theta) + \text{C}_{1}(\triangle_{1})(z_{1} - z - a\theta \cos \theta) \\ &+ \text{K}_{2}(\triangle_{2})(z_{2} - z + b \sin \theta) + \text{C}_{2}(\triangle_{2})(z_{2} - z + b\theta \cos \theta) - \text{Mg} \\ \text{I}\theta &= \text{K}_{1}(\triangle_{1})a(z_{1} - z - a \sin \theta) + \text{C}_{1}(\triangle_{1})a(z_{1} - z - a\theta \cos \theta) \\ &- \text{K}_{2}(\triangle_{2})b(z_{2} - z + b \sin \theta) - \text{C}_{2}(\triangle_{2})b(z_{2} - z + b\theta \cos \theta) \end{split}$$

Forces on front axle (unsprung mass):

$$M_{1}\ddot{z}_{1} = -K_{1}(\Delta_{1})(z_{1} - z - a \sin \theta) - C_{1}(\Delta_{1})(z_{1} - z - a\theta \cos \theta) + \sum_{i=1}^{10} \gamma_{i1}(p_{i1} - z_{1}) - M_{1}g$$

Forces on rear axle (unsprung mass):

$$M_{2}z_{2} = -K_{2}(\Delta_{2})(z_{2} - z + b \sin \theta) - C_{2}(\Delta_{2})(z_{2} - z + b\theta \cos \theta) + \sum_{i=1}^{10} \gamma_{i2}(p_{i2} - z_{2}) - M_{2}g$$

where

$$\Delta_1 = z_1 - z - a \sin \theta , \quad \Delta_1 = z_1 - z - a\theta \cos \theta$$

$$\Delta_2 = z_2 - z + b \sin \theta , \quad \Delta_2 = z_2 - z + b\theta \cos \theta$$

For this study, the suspension spring coefficients were represented by third-order polynomials, as shown in fig. C8. These polynomials were obtained by curve-fitting the actual force-deflection relations, and are reasonable approximations. The suspension damping is as follows:

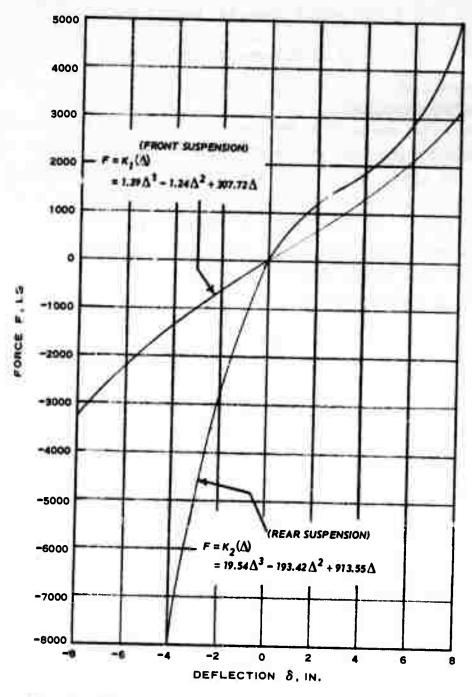


Fig. C8. Force versus deflection for front and rear suspensions of M37 truck model

$$C_1(\triangle_1) = 11.8 \text{ lb-sec/in. (compression)}$$

= 22.8 lb-sec/in. (extension)
 $C_2(\triangle_2) = 12.0 \text{ lb-sec/in. (compression)}$
= 49.0 lb-sec/in. (extension)

Determination of tire-terrain compliance

45. Each wheel, which represented a 9.00x16, 8-PR tire at 45-psi* inflation pressure, was divided into ten segments, five on each side of the tire's center line, as shown in fig. C9. The measured load-deflection relation (fig. C10) for the 9.00x16 tire at 45-psi inflation pressure was such that a center-line deflection of 1.5 in. required a load of 2860 lb. At this deflection, four spring segments are influenced, two on each side of the center line (fig. C9). The spring constant K can be determined from the statics equation:

$$F = \sum_{i=1}^{10} K \cos \emptyset_{i} \triangle_{i}$$

where

Z = vertical displacement of axle

THRESH(i) = height from the zero reference to the ith spring of the undeflected wheel (see fig. C9)

For this case and due to the symmetry of the segments about the center line the equation reduces to

^{*} The difference in the load-deflection relation between 40 psi (at which subsequent tests were conducted) and 45 psi was deemed negligible.

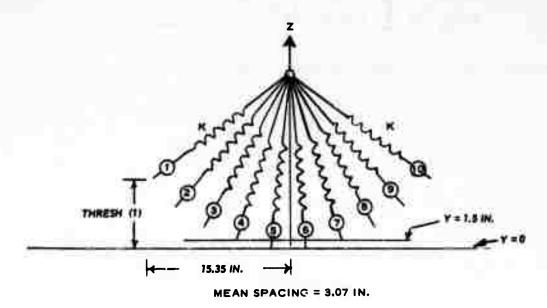


Fig. C9. Schematic of segmented wheel

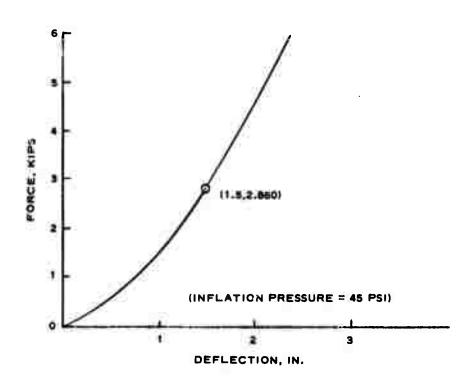


Fig. ClO. Force versus deflection for the 9.00x16, 8-PR tire

2860 = 2K
$$\sum_{i=1}^{2}$$
 1.4 cos $\frac{11.7^{\circ}}{2}$ + 0.7 cos 11.7° + $\frac{11.7^{\circ}}{2}$

where the effective radial deflections are:

$$\Delta_5 = \Delta_6 = 1.4 \text{ in.}$$
 $\Delta_{14} = \Delta_7 = 0.7 \text{ in.}$

Solving for K yields

$$K = 675 lb/in.$$

Defining GAMMA = K $\cos \phi_i = 675 \cos \phi_i$ yields the following relations for the segments of the front and back wheels:

A similar relation is derived for the threshold heights of each segment THRESH(i).

- 46. A mean spacing of 3.07 in. was determined to be adequate for portraying the projected spacing of the springs. As a result, all profile points were spaced 3.07 in. apart, and no interpolation scheme was employed to estimate elevation between adjacent points.
- 47. No damping was incorporated in the tire compliance since in actual vehicles this damping is negligible compared with that of the suspensions. This truck model was forced to traverse each obstacle at a constant speed, and the outputs consisted of motions of the main frame and axles in terms of displacement-, velocity-, and acceleration-time histories.

Computer program

48. A complete listing of the digital computer program and a dictionary of the computer variables used are given at the end of this appendix. This is a FORTRAN IV program written for a GE-430 time-sharing system.

PART III: TEST PROGRAMS

49. Although several series of obstacle-vehicle tests were conducted in efforts to establish resting procedures and criteria for dynamic response, the first test program that yielded data acceptable for verification of the WES vehicle dynamics model was conducted at Aberdeen Proving Ground (APG) with an M60Al tank. These tests were conducted during the period 6-9 May 1968. Subsequently, an obstacle test course was constructed at WES, and tests with an M37 truck were conducted intermittently thereon during 1968 and 1969 for another study. 26

Test Vehicles

50. Pertinent physical characteristics for the M60Al tank and the M37 truck are given in tables C3 and C4, respectively. Photographs of the vehicles are included in fig. C11. The test vehicles were equipped with electronic systems* to measure and record time, distance traveled, and vertical and longitudinal accelerations.

Test Areas

APG

- 51. The test area at APG was a nearly level asphalt strip (fig. C12). Six different sized obstacles were constructed of hardwood and fastened to the strip with steel rods. These obstacles were 12 ft in length and ranged from 6 to 18 in. in height. A sketch of the obstacle configurations with dimensions indicated is shown in fig. C13. WES
- 52. A 200-ft-long test course (fig. C14) constructed of concrete, asphalt, and steel was used for the tests at WES. This course was designed to accommodate either single or multiple obstacles, both rigid and deformable. The obstacles can be arranged to excite primarily the

^{*} This instrumentation is described in detail in reference 20.



a. M60Al tank



b. M37 truck, cargo, 3/4-ton, 4x4

Fig. Cll. Test vehicles



Fig. Cl2. Test course layout, APG

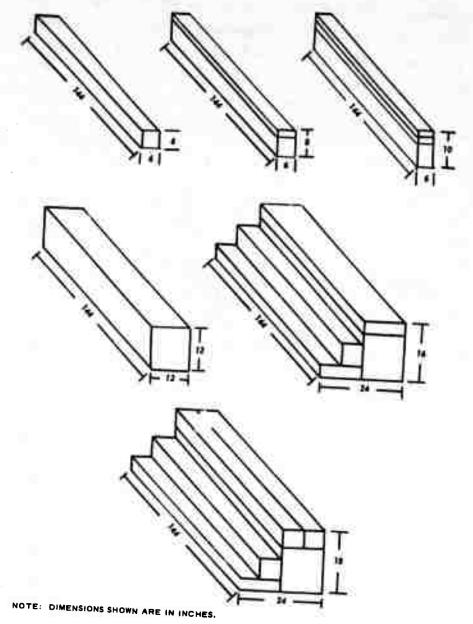


Fig. Cl3. Obstacle configurations



Fig. C14. Obstacle course at WES

pitch and bounce motions of the vehicle (as was done for the tests reported herein) or to excite all components of motion, i.e. pitch, roll, bounce, and yaw, in the direction of and perpendicular to the direction of travel. Only single obstacles were used in this study. The obstacles were half-round in configuration and were welded directly to steel rods embedded in the test course. Obstacle heights were 6, 8, 10, and 12 in.

Tests Conducted

Procedures

- 53. M60Al tank. The vehicle was positioned at right angles to and at a sufficient distance from the obstacle to enable the desired speed to be reached at least 5 ft before striking the obstacle. The test engineer instructed the driver to accelerate to a preselected speed and try to hold this speed constant while crossing the obstacle. Some tests were run at as nearly the same speed as possible over the same obstacle height to determine repeatability of test results.
- 54. M37 truck. The starting position for each test was sufficiently far from the obstacle to permit the driver adequate distance to reach and maintain a constant vehicle speed before striking the obstacle. Several methods of maintaining constant speed while crossing the obstacle were attempted, and the technique that appeared most effective employed an engine tachometer calibrated in miles per hour. Some tests were rerun at the approximate speed of the first test to determine repeatability of test results.

Number of tests

55. The number of tests conducted over selected obstacle heights is given below:

| Obstacle Height | Number of Tests | Conducted |
|-----------------|-----------------|-----------|
| in. | M60Al Tank | M37 Truck |
| 6 | 14 | 10 |
| 8 | 3 | 11 |
| 10 | 4 | 12 |
| , | (Continued) | |

| Obstacle Height in. | Number of Tes | ts Conducted M37 Truck |
|---------------------|---------------|---------------------------|
| 12 16 18 | 11 9 | 2 * |
| Total | 43 | 35 |

^{*} The vertical acceleration was so great on the 12-in.-high obstacle that it was felt tests on higher obstacles might damage the vehicle.

Data Collected

56. For each test, instrumentation installed on the test vehicles recorded continuous measurements of time, distance traveled by the vehicle, and vertical and longitudinal accelerations at selected positions within the vehicle. Event marks on the oscillogram indicated the beginning and end of the test. Impact speed, peak vertical acceleration, and peak longitudinal acceleration were determined from the oscillogram. A summary of these data is given in tables C5 and C6 for the M60Al tank and the M37 truck, respectively. Other data collected included notes, observations, and photographs.

Prediction Data Obtained

57. Predictions of peak vertical acceleration and peak longitudinal acceleration were made for 20 obstacle height-impact speed combinations for the M60Al tank, and predictions of peak vertical acceleration were made for 14 obstacle height-impact speed combinations for the M37 truck, using the previously described models. The speeds selected for the M60Al tank predictions in table C7 were selected on the basis of preliminary reduction of field test data and are generally close, although not identical, to the measured speeds for the M60Al tests shown in table C5. The speeds for the M37 predictions in table C8 were arbitrarily selected to yield dynamic response predictions above and below the 2.5-g peak vertical acceleration tolerance limit.

Method of Analysis and Evaluation of Predictions

- 58. In this study, vehicle performance was described in terms of peak vertical acceleration-speed relations, peak longitudinal acceleration-speed relations, and speed-obstacle height relations for the 2.5- and 2.0-g levels of peak vertical acceleration and peak longitudinal acceleration, respectively. Peak vertical acceleration-speed relations and peak longitudinal acceleration-speed relations for several obstacle heights were developed from measured and predicted data. These relations express the peak vertical acceleration and peak longitudinal acceleration to be expected when a vehicle traverses an obstacle of a specific size at a given speed. From these relations, the speed-obstacle height relations that express the speed at which the vehicle can traverse a given obstacle without exceeding the 2.5-g peak vertical acceleration level and the 2.0-g peak longitudinal acceleration level were developed.
- 59. The prediction accuracy of the models was evaluated by (a) comparing the predicted peak vertical acceleration-speed and peak longitudinal acceleration-speed relations with the measured peak vertical acceleration-speed and peak longitudinal acceleration-speed relations and (b) comparing speeds at 2.5-g peak vertical acceleration and 2.0-g peak longitudinal acceleration levels developed from the predicted relations with the speeds at 2.5-g peak vertical acceleration and 2.0-g peak longitudinal acceleration developed from the measured data. The quality of the prediction accuracy was evaluated on the basis of the simple statistical parameters, root mean square (RMS)* and percent error.**

where Σ = the sum of; n = number of deviations; and d = deviation, i.e. predicted minus measured.

^{*} RMS is a measure of the quality of a relation in terms of the RMS of the deviation

^{**} Percent error = $\frac{\text{predicted - measured}}{\text{measured}} \times 100.$

60. Other conditions peculiar to a particular part of the analysis are discussed in the appropriate section.

Peak Vertical Acceleration-Obstacle Height-Speed Relations

- 61. As stated, vertical acceleration performance was described in terms of peak vertical acceleration-speed relations for several obstacle heights and speed-obstacle height relations for 2.5-g peak vertical acceleration level. Peak vertical acceleration-speed relations developed from measured and predicted data at the driver's seat are shown in plates C1 and C2 for the M60Al tank and M37 truck, respectively. The curves drawn represent the lines of best visual fit. Where the location of the line was doubtful, judgment was aided by the location and curvature of lines on better defined plots. These curves are summarized for the M60Al tank and the M37 truck in plates C3 and C4, respectively. Speed-obstacle height relations for both vehicles are given in plate C5. The speed-obstacle height relations were established from values of speed and corresponding values of obstacle height in plates C3 and C4 at which 2.5-g peak vertical acceleration occurred. M60Al tank
- 62. It can be seen in plates C1 and C3 that both measured and predicted peak vertical accelerations for the M60Al tank at the driver's seat increased with an increase in speed. There appears to be a tendency for the curves to crest at higher speeds, suggesting that after a critical speed has been reached, a further increase in speed would not result in an increase in peak vertical acceleration. In plate C1 it can be seen that except for the 6-in.-high obstacles the predicted peak vertical acceleration was higher than that measured at low speeds, and lower than that measured at higher speeds. The reverse is indicated for the 6-in.-high obstacles. The agreement of the measured and predicted curves for the 8- and 10-in. obstacles is very good, and at the 2.5-g tolerance limit the curves for the 12-, 16-, and 18-in. obstacles seem to be reasonably close.
 - 63. In plate C5, the speed-obstacle height curve for the M6CAl

tank shows that the speed at which 2.5-g peak vertical acceleration is reached at the driver's seat decreases with an increase in obstacle height. The effect of obstacle height on 2.5-g peak vertical acceleration begins to diminish rapidly at about the 9-in. obstacle height. An examination of predicted data points shows that they are in good agreement with curves developed from the measured data.

M37 truck

- 64. In plates C2 and C4, both measured and predicted peak vertical accelerations for the M37 truck at the driver's seat increased with an increase in speed. At the higher speeds the rate of change in peak vertical acceleration was less than that at lower speeds. Except in a few cases at very low speeds (plate C2), there is a large difference between the measured and predicted peak vertical accelerations at the same speed, with the predicted peak vertical acceleration being lower than the measured. An explanation for this is not readily available without further study. It may be because of the difference in the physical characteristics of the vehicle and the model, e.g., spring rates, damping rates, etc., or because of the inability to accurately produce a computer simulation of the motion history of the vehicle immediately prior to and during obstacle traversal.
- 65. The speed-obstacle height curve for the M37 truck in plate C5 was established from the measured relations for the 6-, 8-, and 10-in.-high obstacles. (Note in plate C2 that even at low speeds the peak vertical acceleration measured over 12-in.-high obstacles greatly exceeded 2.5 g's.) The relation is similar to that discussed above for the M60Al tank except that 2.5-g peak vertical acceleration at the driver's seat was reached at lower speeds and obstacle heights than that for the M60Al tank. It can be seen that the predictions indicate a higher speed for crossing 8- and 10-in.-high obstacles than that shown by the relations developed from measured data. In addition, the predicted speed for the 12-in.-high obstacle was 4 mph, while the measured data indicate (see again plate C2) that the vehicle cannot cross a 12-in.-high obstacle at any speed without exceeding 2.5-g peak vertical acceleration. No predicted value is shown for 6-in.-high obstacles

because the predictions of peak vertical acceleration at the driver's seat did not reach 2.5 g at speeds up to 20 mph. Hence, it is apparent that for all obstacle heights tested, the predicted speed at which 2.5-g peak vertical acceleration at the driver's seat would occur was greater than that indicated by the relations developed from measured data. Prediction accuracy

- 66. The measured and predicted peak vertical accelerations, the deviations, and the percent error for the speeds at which comparisons were made are given in tables C7 and C8 for the M60Al tank and the M37 truck, respectively. The measured data shown in column 4 of tables C7 and C8 were obtained from the relations established for measured data shown in fig. a, plate C3, for the M60Al tank and in fig. a, plate C4, for the M37 truck. The predicted data given in column 3 of tables C7 and C8 are for predictions made for the speeds shown in column 2.
- 67. The measured and predicted peak vertical acceleration data given in tables C7 and C8 are plotted in plate C6 for both vehicles, and a summary of the RMS and average percent error for each obstacle height is given in the following tabulation:

| | M60 <i>i</i> | Al Tank | _M37 | Truck |
|---------------------------|--------------|-----------------------------|--------------|-----------------------------|
| Obstacle Height in. | RMS | Average Percent Error | RMS | Average Percent Error |
| 6 8 | 0.59 0.10 | 81 | 2.06 | 65 |
| 10 | 0.22 | 11 41 | 1.45 1.17 | 28 18 |
| 12 16 | 0.43 0.87 | 18 22 | 4.00 | 68 |
| 18 | 0.16 | 6 | | |
| Overall | 0.51 | 37 | 1.94 | 39 |

In the tabulation above it can be seen that the overall percent error for the tracked vehicle was only slightly lower than for the wheeled vehicle, but in terms of RMS the predictions for the tracked vehicle were much better than for the wheeled vehicle. For each vehicle at several obstacle heights the percent error and RMS are sufficiently large to indicate that some refinement is needed in the mathematical

models. In fig. a, plate C6, the predictions for the M60Al tank are generally reasonable except for a few points—the two highest accelerations for the 6- and 16-in. obstacles, and the highest acceleration for the 12-in. obstacle. In fig. b, plate C6, the predictions for the M37 truck are generally poor except for three points—the two lowest accelerations for the 10-in. obstacle and the lowest acceleration for the 8-in. obstacle.

68. A comparison of measured and predicted speeds at 2.5-g peak vertical acceleration was made by selecting the speed and obstacle heights at which 2.5-g peak vertical acceleration occurred in the figures given in plates C3 and C4. The data obtained in this manner, along with values of prediction accuracy, are given in the following tabulation.

| | | M60A1 | Tank | | | M37_ | Truck | | | |
|---------------------------|------|--------------------------------|--|------------------|---|------|-------------------|-------|-------------------------------------|------------------|
| Obstacle Height in. | | mph, at g PVA* Predicted | Deviation (Predicted - Measured) | Percent Error | Speed, mph, at 2.5-g PVA Measured Predicted | | Percent 2.5-g PVA | | Deviation (Predicted easured) | Percent Error |
| 6 | | | | | 8.8 | | | | | |
| 8 | 15.6 | 15.6 | 0.0 | 0 | 5.1 | 7.3 | 2.2 | 35 | | |
| 10 | 11.0 | 11.€ | 0.6 | 6 | 11.14 | 5.0 | 0.6 | 14 | | |
| 12 | 7.4 | 7.2 | -0.2 | 3 | | 4.0 | | | | |
| 16 | 5.3 | 6.1 | 0.8 | 15 | | | | | | |
| 18 | 4.9 | 5.1 | 0.2 | žĻ | | •• | | | | |
| | | | Avera | gc 7 | | | Averag | ge 25 | | |

^{*} IVA designates peak vertical acceleration.

The speed-obstacle height relations developed from the data in the tabulation above are shown in plate C5. It can be seen that the relations are reasonably well defined and that as obstacle height increases, the speed at which the limiting acceleration value is reached decreases and that the predicted data agree reasonably well with the curves drawn for the measured data, particularly for the M60Al tank. For a given obstacle height, the M60Al tank can cross that obstacle at almost twice the speed that the M37 truck can cross it before 2.5-g peak vertical acceleration is reached. The tabulation above and plate C7 reveal that the accuracy of predicting speeds at 2.5-g vertical accleration is very good for the M60Al tank and reasonably good for the M37 truck throughout the speed range shown.

Teak Longitudinal Acceleration-Obstacle Height-Speed Relations

69. Longitudinal acceleration performance was described in terms of peak longitudinal acceleration-speed relations for several obstacle heights, and speed-obstacle height relations for 2.0-g peak longitudinal acceleration. Peak longitudinal acceleration-speed relations developed from measured and predicted data at the driver's seat are shown in plate C8 for the M60Al tank. Since no predictions of peak longitudinal acceleration were made for the M37 truck, the peak longitudinal acceleration-speed relations from measured data only are shown in plate C9. The curves drawn represent the lines of best visual fit, and they are summarized in plates C9 and C10 for the M37 truck and the M60Al tank, respectively. Speed-obstacle height relations for the M60Al tank are shown in plate C11. The speed-obstacle height relation for the M60Al tank was established from values of speed and corresponding values of obstacle height in plate C10 at which 2.0-g peak longitudinal acceleration occurred.

M60Al tank

- 70. It can be seen in plates C8 and C10 that both measured and predicted peak longitudinal accelerations for the M60Al tank at the driver's seat increased with an increase in speed. In all cases, the predicted curves indicate higher values of peak longitudinal acceleration than the measured curves. Although the scatter in data is not excessive, additional data are required to better define the relations. In table C5 it can be seen that the cest data include only one measured peak longitudinal acceleration greater than 2.0 g's. However, the curves for the larger obstacles shown in plates C8 and C10 were extended to 2.0 g's.
- 71. The speed-obstacle height relations (plate Cll) for the M60Al tank show that the speed at which 2.0-g peak longitudinal acceleration is reached at the driver's seat decreases with an increase in obstacle height. The predicted data shown in the plot do not agree very well with the measured data.

M37 truck

72. The curves in plate C9, developed for measured data only, indicate that the peak longitudinal acceleration of the M37 truck increased with speed up to about 8 to 10 mph, and then decreased with further increase in speed. An increase in peak longitudinal acceleration with increased obstacle height is also indicated, as would be expected. While the scatter of the data points is not excessive, there is an obvious need for additional data, especially for the 12-in. obstacle height.

Prediction accuracy

- 73. The quality of the prediction accuracy for peak longitudinal acceleration-speed and obstacle height-speed relations for the M6OAl tank was determined in the same manner as that for the vertical acceleration relations. The measured and predicted data, deviation, and percent errors for the speeds at which comparisons were made are given in table C7. The measured data shown in column 8 of table C7 were obtained from the relations established from measured data shown in plate C10. The predicted data shown in column 7 are for predictions made for the speeds shown in column 2.
- 74. The measured and predicted peak longitudinal acceleration data given in table C7 for the M60Al tank are plotted in plate C12, and a summary of the RM3 and average percent error for each obstacle height is given below.

| | _M60A | l Tank |
|----------|-------|---------|
| Obstacle | | Average |
| Height | | Percent |
| in. | RMS | Error |
| 6 | 0.:26 | 74 |
| 8 | 0.33 | 80 |
| 10 | 0.22 | 43 |
| 12 | 0.56 | 102 |
| 16 | 1.07 | 119 |
| 18 | 1.05 | 76 |
| | | |
| Overall | 0.64 | 82 |

The preceding tabulation shows that the overall prediction accuracy is not very good and no consistent pattern is evident other than that prediction accuracy generally decreased with obstacle height increase. In plate C12 the individual data points reveal perhaps more meaningful trends. Except for two points for the 6-in. obstacle height, predicted peak longitudinal accelerations are higher than the measured accelerations. Predictions for the 12-, 16-, and 18-in.-high obstacles are consistently higher than the measured data. The deviations for these obstacle heights are also much greater than for the other obstacle heights.

75. A comparison of M6OAl measured and predicted speeds at 2.0-g peak longitudinal acceleration was made by selecting the speed and obstacle height at which 2.0-g peak longitudinal acceleration occurred in plate ClO. Since only two points were available for the 2.0-g level of acceleration, a comparison was also made for the 1.0-g level of peak longitudinal acceleration. The data obtained in this manner, along with values of prediction accuracy, are given below.

| | | | | M60A1 | Tank | | | |
|---------------------------|----------------------------|--------------------------------|--|---------------------|------|--|---------|-------|
| Obstacle Height in. | epeed, 2.0- Measured | mph, at g PLA* Predicted | Deviation (Predicted - Measured) | Percent 1.0-g PIA (| | Deviation (Predicted - Measured) | Percent | |
| 6 | | | | | 17.3 | | | |
| 8 | | *- | | | 14.5 | 12.0 | -2.5 | -17 |
| 10 | | | ~= | | 12.0 | 10.6 | -1.4 | -12 |
| 12 | | | | | 8.0 | 4.2 | -3.8 | -1,3 |
| 16 | 10.0 | 5.6 | -14.14 | -44 | 6.6 | | | |
| 18 | 7.4 | 4.6 | -2.8 | - 38 | 4.6 | | | |
| | | | Average | -41 | | | Average | e -26 |

^{*} PLA designates peak longitudinal acceleration.

The speed-costacle height relations developed from the data given in the tabulation above for the M60Al tank are shown in plate Cll. The data from which the curve for the speed-obstacle heights at which 2.0-g peak longitudinal acceleration is reached are limited, but the curve for the 1.0-g level is well defined. Both curves show that as the obstacle height increases, the speed at which the limiting acceleration value is reached decreases. From the data above and plate Cl3, it can be

seen that the predicted speed at which a given level of peak longitudinal acceleration will occur for a given obstacle height is lower than the measured speed. The difference in measured and predicted values is greater for the 2.0-g level than for the 1.0-g peak longitudinal acceleration level. The prediction accuracy for the 1.0-g peak longitudinal acceleration level is acceptable for the 8- and 10-in. obstacle heights, but it is not acceptable for the 12-in. obstacle. For the 2.0-g peak longitudinal acceleration level, the prediction accuracy is not acceptable.

Notes and Observations

76. During the test program, it was observed that in all of the tests the vehicles appeared to strike the obstacles at or very near an angle of 90 deg; however, in some tests there was a change in vehicle orientation during traversal of the obstacle. Since the peak vertical acceleration generally occurs when the vehicle strikes the ground after crossing the obstacle, the magnitude of the peak vertical acceleration is influenced by the attitude and orientation of the vehicle. The M60Al, for instance, might strike the ground with both tracks simultaneously, thus deriving the maximum benefit of the suspension system; one track may strike the ground first with the vehicle in such position that the major portion of the shock is imposed on the suspension system of one track only, resulting in some roll motion; or the vehicle may strike the ground in any position between these two extremes with a possibility of a slight change in vehicle direction. The extremes are even wider for the M37 truck. A wheeled vehicle may land on all wheels simultaneously, on two wheels, or on a single wheel. Not all of these extremes were evidenced in these tests; however, there were occasions when one trom or wheel appeared to make contact with the ground before the other track or wheel.

77. In the M60Al tests, the peak longitudinal acceleration at the driver's seat occurred when the vehicle contacted the obstacle, whereas the peak longitudinal acceleration of the M37 truck generally

occurred when the front wheels struck the obstacle, although there were some occasions when the peak longitudinal acceleration occurred when the rear wheels contacted the obstacle. Since some combinations of obstacle height and speed caused the M37 truck to become airborne, i.e. lose contact with the ground, it is easy to hypothesize a condition in which the peak longitudinal acceleration for the M37 might occur when the vehicle strikes the ground after crossing the obstacle. The M60Al tank did not become airborne during any of the tests described herein.

78. As previously stated, the drivers attempted to maintain a constant speed across the obstacle. Since this would be patently impossible if the driver waited until he felt the vehicle slowing before he applied additional power, the procedure evolved was to apply additional power as close to the moment of impact as possible. The success of this procedure, in terms of repeatability of the test results, has its limitations. Obviously, there was nothing the driver could do toward maintaining a constant speed when the M37 truck was airborne. Rather than attempting to maintain a constant speed, it is suggested that future tests be conducted with the throttle in a fixed position throughout the test.

PART V: CONCLUSIONS AND RECOMMENDATIONS

Conclusions

- 79. Based on the data reported herein and subject to the limits imposed by these data, the following conclusions are offered:
 - a. Discrete rigid obstacles affect vehicle performance by producing adverse vertical and longitudinal motions that may endanger the driver or damage the cargo.
 - b. The magnitudes of peak vertical and longitudinal accelerations are dependent primarily on speed, obstacle height, and characteristics of the vehicle suspension system.
 - c. Vehicle performance in terms of peak vertical and peak longitudinal accelerations when traversing discrete, rigid obstacles can be correlated with impact speed.
 - d. The speed at which the MoOAl tank may contact a rigid obstacle without exceeding the 2.5-g peak vertical acceleration tolerance limit or the 2.0-g peak longitudinal acceleration tolerance limit can be predicted by the described mathematical techniques with reasonable accuracy.
 - e. The speed at which the M37 may contact a rigid obstacle without exceeding the 2.5-g peak vertical acceleration tolerance limit can be predicted by the described mathematical techniques with reasonable accuracy for 8- to 10-in.-high obstacles. Above and below this height range, however, the prediction accuracy left much to be desired.
 - f. Refinement is needed in the dynamic response prediction models with more improvement required in predicting dynamic response of wheeled vehicles.

Recommendations

80. It is recommended that:

- a. Additional systematic controlled testing be done with tracked and wheeled vehicles to refine and extend the relations and revise the vehicle dynamic prediction models presented herein.
- <u>b</u>. Refinements be made in the dynamic response prediction models to bring the predictions, particularly those for

- higher speed and acceleration levels, more clearly in line with the measured aynamic response.
- c. Investigation be continued to explore, in addition to peak acceleration, other performance parameters such as root mean square acceleration and absorbed power as tolerance descriptors.

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Table Cl
Vehicle Characteristics Used in Predicting M60Al Dynamic Response

| Distance from road whe | el to body o | center of grav | ity: |
|--|--------------|----------------|-------------------------------|
| Road wheel ℓ_1 . | | | 76 in. |
| Road wheel ℓ_2 . | | | 44 in. |
| Road wheel ℓ_3 . | | | 12 in. |
| Road wheel ℓ_4 . | | | 24 in. |
| Road wheel ℓ_5 . | | | 56 in. |
| Road wheel 16. | • • • • • • | | 88 in. |
| Body pitch inertia . | | | $581,700 \text{ inlb/sec}^2*$ |
| Sprung weight | | • • • • • • | 54,500 1b* |
| Vertical distance S to body center of gr | | | n 46 in. |
| Effective unsprung wei | | | |
| lower track assembly | | | 1420 lb |

^{*} The values for the body pitch inertia and sprung weight represent but one-half of the true value in order to fit the two-dimensional model which represents one-half of the vehicle that is assumed to be split down its longitudinal axis.

Table C2

Vehicle Characteristics Used in the Dynamic Response Predictions for the M37 Truck

| Vertical distance from wheel hub center to body center of gravity: |
|--|
| Road wheel ℓ_1 |
| Road wheel ℓ_2 |
| Road wheel ℓ_3 |
| Road wheel ℓ_{1} |
| Undeflected wheel radius |
| Total weight (loaded) |
| Sprung weight |
| Unsprung weight |
| Front axle |
| Rear axle |
| Sprung* pitch moment of inertia about center of gravity |
| Sprung roll moment of inertia about |
| center of gravity |
| Front axle roll moment of inertia 1256 lb-sec ² /in. |
| Rear axle roll moment of inertia |
| Wheel travel from static to bump stop: |
| Front 4 in. |
| Rear |
| Tire (9.00-16, 8-PR, 45 psi) damping rate 9.56 lb-sec/in. |

Note: One-half of these values was used in the two-dimensional model under the assumption that the vehicle is symmetrical.

^{*} Sprung mass, $M = \frac{6212}{386} = 16.1 \text{ lb-sec}^2/\text{in}$. Front axle unsprung mass, $M_1 = \frac{722}{386} = 1.87 \text{ lb-sec}^2/\text{in}$. Rear axle unsprung mass, $M_2 = \frac{616}{386} = 1.60 \text{ lb-sec}^2/\text{in}$.

Table C3

Summary of Vehicle Characteristics and Performance Data for M60Al Tank

| Cross-country gross weight, fully equipped plus payload and personnel |
|---|
| Track weight, 1b Left |
| Dimensions |
| Overall length, in. (hull only) |
| Vehicle approach angle, deg |
| Vehicle departure angle, deg |
| Ground clearance of hull between tracks, in |
| Force leading edge can withstand, lb Not limite |
| Winch capacity, 1b |
| Water performance characteristics |
| Fording depth Normal fording (no kit), in |
| Engine 910. |
| Make |
| Transmission |
| Make |
| lst gear |
| |

(Continued)

Table C3 (Concluded)

Table C4

Summary of Vehicle Characteristics and Performance Data for the M37 Truck

| | | | | | _ | | | | | |
|-----------------------------------|---|-------|-----|-----|----|---|----|---|-----|---|
| Cross-c | country gross weight, fully equipped plonnel | lus | pay | ylo | ad | a | nd | | | |
| T | Axle loads (front to rear), lb No. 1 No. 2 Total gross weight, lb Payload, lb (cross country) | • | • | | • | • | • | • | ٠ | 3,600 3,600 7,200 1,500 |
| Dimensi | ions, in. | | | | | | | | | |
| W W D | Overall length (including winch if avaiumellase | • | • • | • | • | • | • | | • | 189.4 112 26 72.8 62 112 |
| | approach angle, deg | | | | | | | | | 38 |
| | departure angle, deg | | | | | | | | | 32 |
| | rriage clearance, in. | | | | | | | | | |
| 41 | xle | • | • • | • | • | • | • | • | • | 11 16 |
| Force le | eading edge can withstand, lb | • , | | • | | | | | | 10,000 |
| | apacity, lb | | | | | | | | | 7,500 |
| | erformance characteristics | | | | | | | | | |
| | ording depth, in. Normal | | | | | | | | | 30 79 2 |
| | No. | | | | | | | | | |
| Mo Fu Br Ma En | ake odel call type cake horsepower. aximum torque, lb-ft ngine RPM at maximum torque ngine RPM at brake horsepower. | • • • | • | • | • | | • | | • (| Dodge T-245 Gasoline 77 188 1,200 3,200 |

(Continued)

Table C4 (Concluded)

| Transmission | |
|---|---------------------|
| Type or model | r-245 -3 955 |
| Ratios | 6.40:1 |
| 2d | 3.09:1 |
| 3d | 1.69:1 |
| 4th | 1.00:1 |
| Transfer case | |
| Model | Timken |
| Ratios | |
| High | 1.00:1 |
| Low | 1.96:1 |
| Axles | |
| Model | T-245 |
| Ratio | 5.83:1 |
| Tire data | |
| Туре | d and snow |
| Size | 9.00x16 |
| Ply | 8 |
| Tread design | NDMS |
| Unloaded diameter (including tread), in | 25 |
| Unloaded width, in. | 10.2 |
| Tread depth, in | į |
| No. of tires | 4 |
| No. of wheels | 4 |
| Cross-country inflation pressure, psi | 14 |
| Highway inflation pressure, psi | 40 |
| Steering data | |
| Turning radius (curb to curb), ft | 25 |
| Time required to steer from straight | 2) |
| ahead to full lock turn, sec | 3 |
| Distance from front wheel steering (hub) | 3 |
| pivot to front of vehicle, in | 35.62 |
| Distance from front wheel steering (hub) | |
| pivot to outside of vehicle, in | 10.0 |
| Center of gravity location, in. | |
| Horizontal distance from front axle | 63 |
| Vertical distance above ground at full load | |
| static position | 34 |
| | |

Table C5

Summary of Measured Data and Test Results for M60Al Tank Tests

| Test | Field Identification No. | Obscacle Height in. | Impact Speed mph | Peak Vertical Acceleration g's | Peak Longitudinal Acceleration g's |
|---------------------------------|--------------------------------|----------------------------|------------------------|--------------------------------------|------------------------------------|
| 1 2 3 | 1-1 1-2 | 6 | 3.75 4.87 | 0.42 0.42 | 0.24 0.21 |
| 2 3 4 5 6 7 8 | 1-3 2-1 | 6 6 6 6 6 6 | 3.52 6.20 | 0.42 0.48 | 0.12 0.27 |
| 6 | 2 - 2 2 - 3 | 6 | 5.88 5.81 | 0.54 0.54 | 0.42 |
| 7 | 3-1 | 6 | 10.51 | 0.48 | 0.36 0.36 |
| | 3-2 | 6 | 10.27 | 0.48 | 0.33 |
| 9 10 | 3 - 3 4-1 | 6 | 10.05 | 0.54 | 0.39 |
| 11 | 4-2 | 6 6 | 16.50 17.17 | 0.73 | 0.89 |
| 12 | 4-3 | 6 | 17.47 | 0.54 0.54 | 0.89 0.89 |
| 13 | 5-2 | 6 6 | 18.15 | 0.67 | 1.13 |
| 14 | 5 - 3 | 6 | 18.15 | 0.48 | 1.12 |
| 15 16 | 15 - 1 15 - 2 | 8 8 | 6.15 | 0.50 | 0.47 |
| 17 | 15-3 | 8 | 11.25 17.62 | 1.56 2.75 | 0.38 1.47 |
| 18 | 16-1 | 10 | 6.07 | 0.62 | 0.38 |
| 19 20 | 16-2 16-3 | 10 10 | 10.65 | 2.44 | 0.94 |
| 21 | 16-4 | 10 | 17.17 17.27 | 3.12 2.56 | 1.44 1.26 |
| 55 | 6-1 | 12 | 4.42 | 0.96 | 0.36 |
| 23 24 | 6 - 2 6 - 3 | 12 12 | 5.02 | 1.39 | 0.89 |
| 25 | 7-1 | 12 | 4.87 6.45 | 1.82 1.88 | 0.36 |
| 26 | 7-2 | 12 | 5.85 | 1.51 | 0.60 0.36 |
| 2 7 28 | 7-3 | 12 | 6.07 | 1.82 | 0.30 |
| 29 | 9-1 9-2 | 12 12 | 8.44 8.60 | 4.24 | 1.49 |
| 30 | 9 - 3 | 12 | 8.60 | 3.64 3.82 | 1.19 |
| 31 | 8-1 | 12 | 10.17 | 2 .7 2 | 1.25 1.88 |
| 32 | 8-2 | 12 | 10.52 | 3.76 | 1.40 |
| 33 34 | 10 - 1 10 - 2 | 16 | 5.40 | 2.56 | 0.73 |
| 35 | 10-3 | 16 16 | 4.50 4.42 | 1.44 | 1.03 |
| 36 | 11-1 | 16 | 6.37 | 1.56 3.75 | 1.76 |
| 37 38 | 11-2 | 16 | 6.91 | 3.44 | 0.91 1.18 |
| 38 30 | 11-3 | 16 | 6.52 | 4.25 | 1.03 |
| 39 40 | 12-1 12-2 | 16 16 | 8.7, | 3.75 | 2.29 |
| 41 | 12-3 | 16 | 7.65 8.77 | 3.75 3.12 | 1.32 1.41 |
| 42 | 13-2 | 18 | 5.40 | 2.87 | 1.18 |
| 43 | 13-3 | 18 | 4.92 | 2.50 | 1.20 |

Table C6
Summary of Measured Data for M37 Truck Tests

| Test | Field Identification No. | Obstacle Height in. | Impact Speed mph | Peak Vertical Acceleration g's | Peak Longitudinal Acceleration g's |
|------------------|--------------------------------|---------------------------|------------------------|--------------------------------------|--|
| 44 | F68-0024 | 6 | 4.0 | 1.48 | 0.74 |
| 45 | F68-0025 | 6 | 3.5 | 1.43 | 0.76 |
| 46 | F68-0026 | 6 | 6.4 | 1.48 | 0.82 |
| 47 | F68-0027 | 6 | 8.9 | 2.57 | NA |
| 48 | F68-0028 | 6 | 11.8 | 3.40 | 0.80 |
| 49 | F68-0029 | 6 | 11.6 | 3.22 | 0.76 |
| 50 | F68-0030 | 6 | 15.1 | 3.66 | 0.79 |
| 51 | F68-0031 | 6 | 14.5 | 3.40 | 0.82 |
| 52 | F68-0032 | 6 | 15.3 | 3.85 | 0.70 |
| 53 | F68-0033 | 6 | 14.8 | 3.64 | 0.81 |
| 54 | F68-0103 | 8 | 2.8 | 1.64 | 0.85 |
| 55 | F68-0104 | 8 | 3.0 | 1.63 | 0.96 |
| 56 | F68-0105 F68-0106 | 8 | 8.1 | 4.38 | 1.57 |
| 5 7 58 | F68-0107 | 8 8 | 5.4 | 2.43 | 1.50 |
| 59 | F68-0109 | Ο Ω | 10.8 | 5.78 | 1.48 |
| 60 | F68-0110 | 8 8 8 8 | 13.3 | 6.59 | 1.45 |
| 61 | F68-0111 | 8 | 13.3 19.1 | 6 . 20 7 . 99 | 1.50 |
| 62 | F68-0112 | 8 | 9.3 | 7.99 3.44 | 1.19 1.48 |
| 63 | F68-0113 | 8 | 19.0 | 8.73 | 1.40 |
| 64 | F68-0114 | 8 | 19.6 | 8.72 | 1.40 |
| 65 | F68-0034 | 10 | 2.9 | 1.68 | 1.07 |
| 66 | F68-0035 | 10 | 3.0 | 1.77 | 1.05 |
| 67 | F68-0036 | 10 | 2.3 | 1.59 | 1.11 |
| 68 | F68-0037 | 10 | 6.2 | 4.56 | 1.64 |
| 69 | F68-0038 | 10 | 10.3 | 6.47 | 1.56 |
| 70 | F68-0039 | 10 | 6.9 | 4.58 | 1.64 |
| 71 72 | F68-0040 F68-0041 | 10 | 10.5 | 7.45 | 1.84 |
| 73 | F68-0042 | 10 | 10.6 | 5.85 | 1.71 |
| 74 | F68-0043 | 10 10 | 15.2 | 7.30 | 1.42 |
| 7 5 | F68-0044 | 10 | 15.8 15.5 | 9.00 | 1.76 |
| 76 | F68-0045 | 10 | 11.5 | 8.39 7.67 | 1.41 1.69 |
| 77 | F68-0046 | 12 | 3.3 | 6.29 | 1.70 |
| 78 | F68-0047 | 12 | 3.4 | 6.43 | 1.30 |

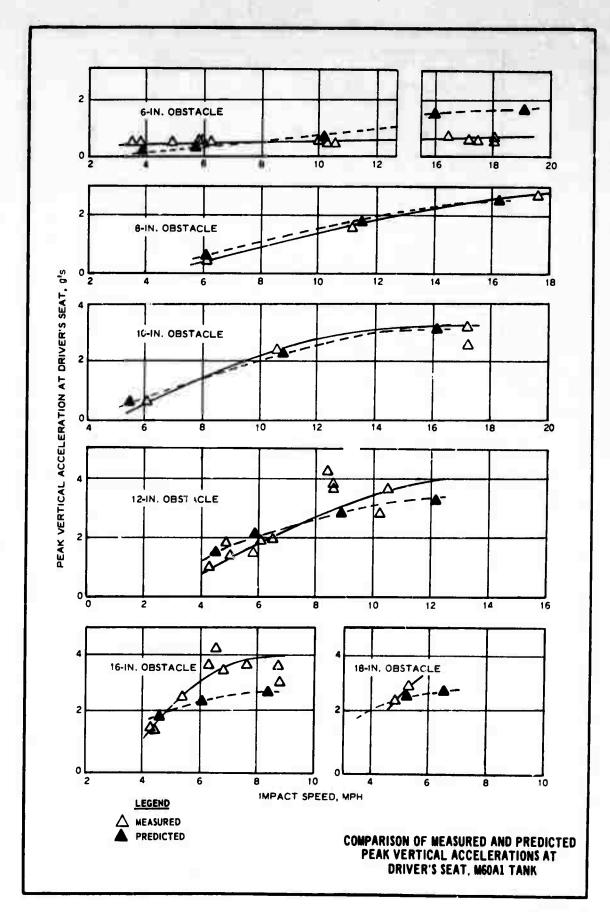
Table C7

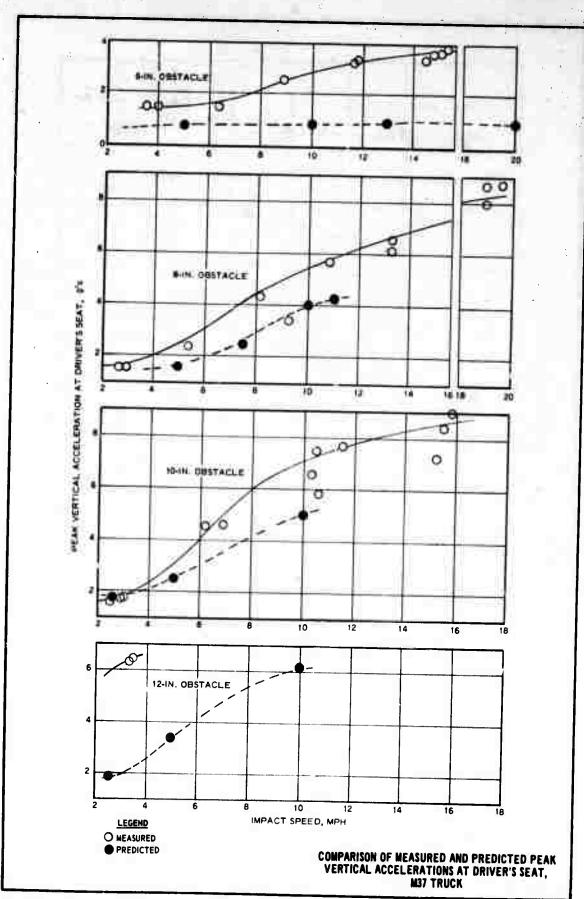
Summary of Predicted Data and Comparison of Prediction Accuracy for M60Al Tank

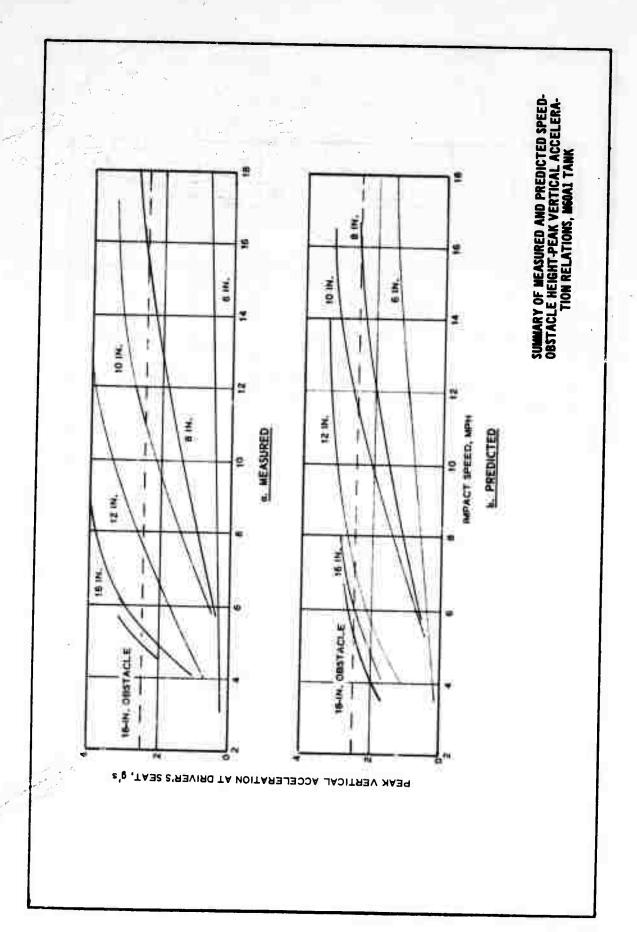
| lon | Percent Error | 105 | 100 75 70 70 70 70 70 70 70 70 70 70 70 70 70 | 174 65 0 | 103 11 11 | 227 162 10 9 | 139 139 78 | 8E |
|---|------------------|---------------------------------------|---|------------------------|------------------------|-------------------------------|------------------------|-------|
| Seat 1 Accelerat | Deviation g : | 0.21 | 0.36 0.05 0.25 | 0.39 0.00 | 0.31 0.12 0.16 | 0.75 0.81 0.13 0.15 | 1.04 | 1.00 |
| Driver's Seat Peak Longitudinal Acceleration | Measured g's | 0.20 | 0.36 0.78 1.15 | 0.23 0.60 1.24 | 0.30 0.88 1.46 | 0.33 0.50 1.26 1.70 | 0.65 0.88 1.60 | 1.22 |
| Peak | Predicted g's | · · · · · · · · · · · · · · · · · · · | 0.72 0.83 0.90 | 0.63 0.99 1.24 | 0.61 | 1.08 1.31 1.39 1.85 | 1.69 2.10 2.72 | 2.22 |
| | Percent Error | -50 -37 | 134 134 | % - 1 % | 071 | 37. 9 -9 -18 | 11 -20 -34 | ۹ : |
| Seat Acceleration | Deviation g's | -0.20 | 0.19 C.87 0.94 | 0.15 0.02 -0.07 | 0.36 0.10 0.01 | 0.39 0.18 0.28 | 0.20 -0.63 -1.36 | -0.16 |
| Driver's Peak Vertical | Measured g's | 0.40 | 0.44 0.61 0.70 | 0.50 1.81 2.63 | 0.30 3.20 | 1.05 1.92 3.10 4.00 | 1.78 3.12 4.05 | 2.85 |
| Pea | Predicted g's | 0.20 | 0.63 1.48 1.64 | 0.65 | 0.66 2.30 3.19 | 1.44 2.10 2.82 3.30 | 1.98 2.49 2.69 | 2.66 |
| Impact | Speed | 3.90 | 10.15 15.97 19.17 | 6.12 11.56 16.32 | 5.44 10.88 16.09 | 4.45 5.89 8.84 12.24 | 4.64 6.12 8.38 | 5.27 |
| Obstacle | Height in. | 600 | ००० | ∞∞∞ | 10 10 10 | ឧឧឧឧ | 16 16 16 | 18 |

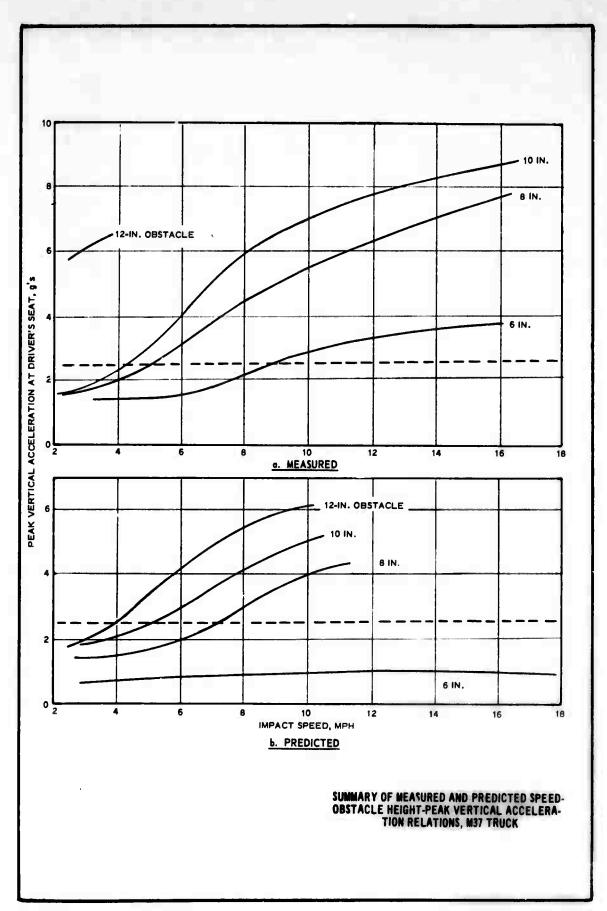
Table C8
Summary of Predicted Data and Comparison of Prediction
Accuracy for M37 Truck

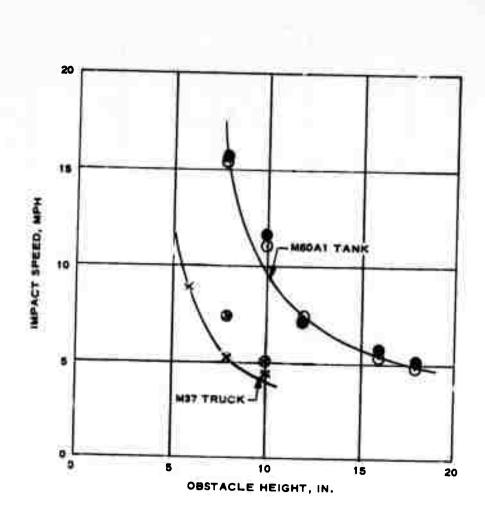
| Obstacle | Impact | Driver's Seat Peak Vertical Acceleration | | | | | | |
|----------------|-----------------------------|--|------------------------------|----------------------------------|--------------------------|--|--|--|
| Height in. | Speed mph | Predicted g's | Measured g's | Deviation g's | Percent Error | | | |
| 6 6 6 | 5.0 10.0 13.0 20.0 | 0.70 0.85 0.94 0.85 | 1.40 2.98 3.72 | -0.70 -2.13 -2.78 | -50 -71 -75 | | | |
| 8 8 8 | 5.0 7.5 10.0 11.0 | 1.41 2.43 4.00 4.29 | 1.64 4.10 5.70 5.92 | -0.23 -1.67 -1.70 -1.63 | -14 -41 -30 -28 | | | |
| 10 10 10 | 2.5 5.0 10.0 | 1.78 2.52 5.03 | 1.60 2.94 7.00 | 0.18 -0.42 -1.97 | 11 -14 -28 | | | |
| 12 12 12 | 2.5 5.0 10.0 | 1.90 3.40 6.09 | 5.90 | -4.00 | -68 | | | |







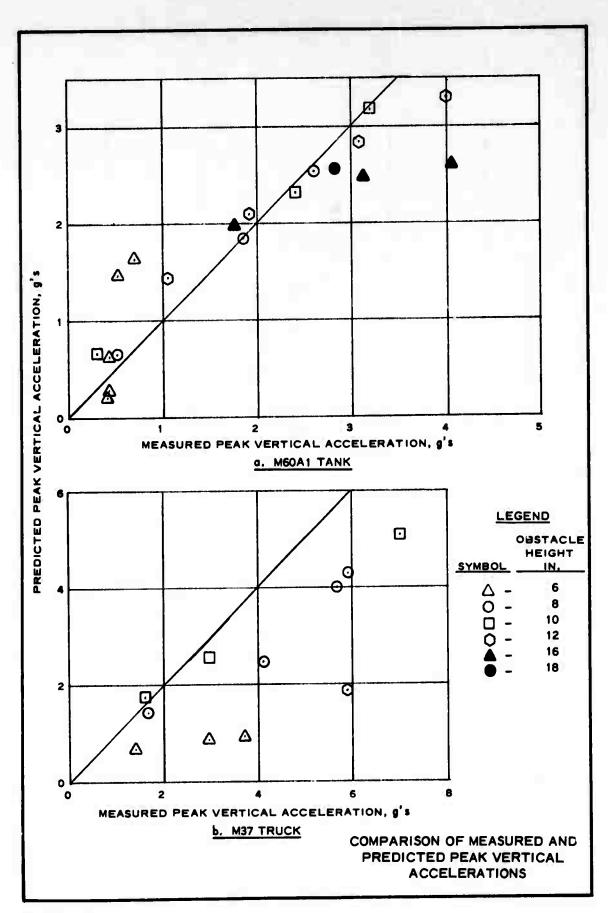


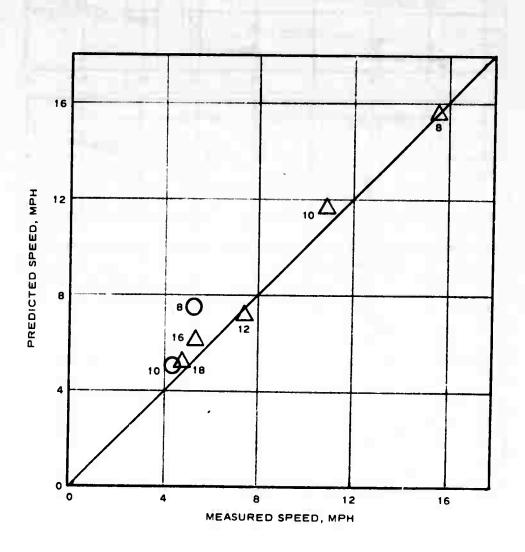


LEGEND

- O MEASURED, MOOAT TANK
- PREDICTED, M60A1 TANK
- X MEASURED, M37 TRUCK
- Ø PREDICTED, M37 TRUCK

SPEED-OBSTACLE HEIGHT RELA-TIONS AT 2.5-9 PEAK VERTICAL ACCELERATION



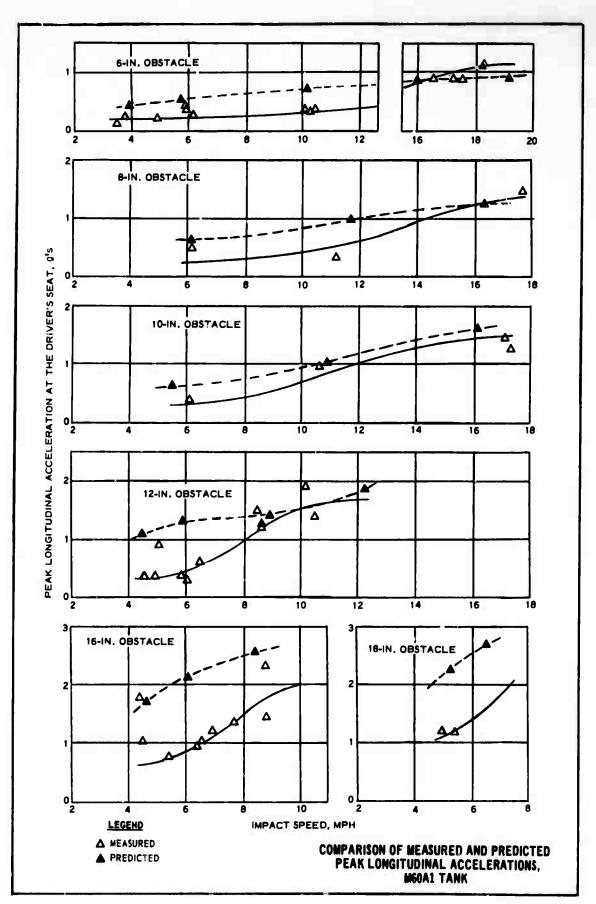


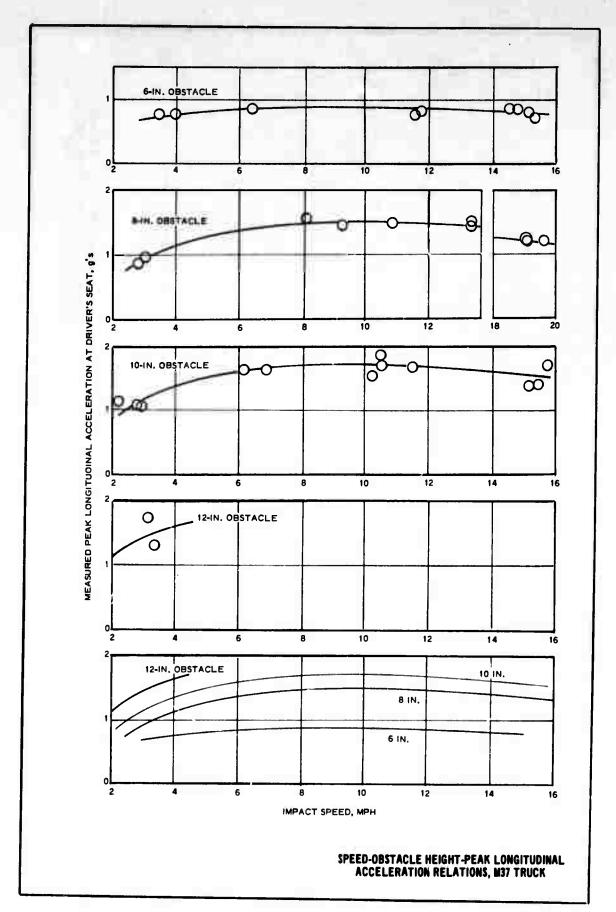
NOTE: NUMBERS NEAR PLOTTED POINTS INDI-CATE OBSTACLE HEIGHT IN INCHES.

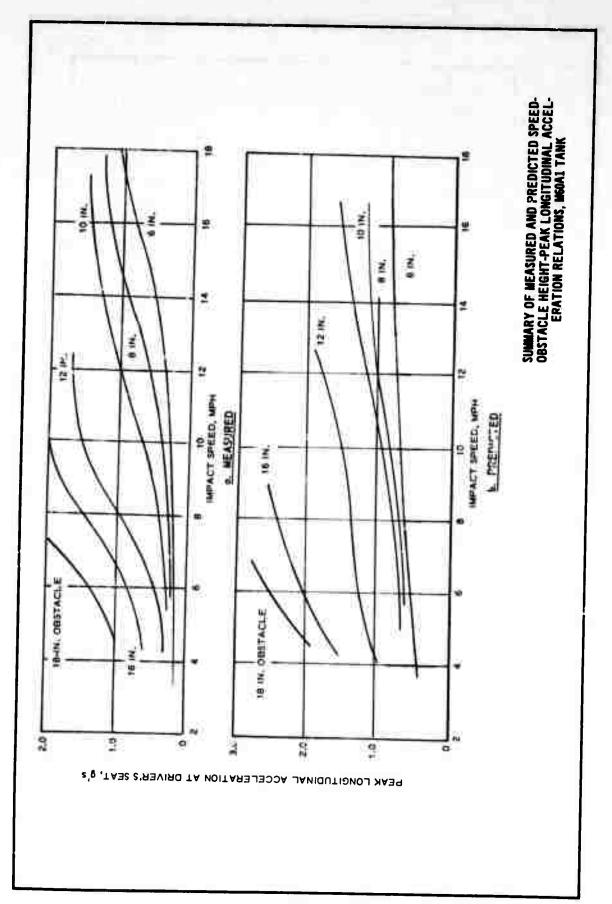
LEGEND

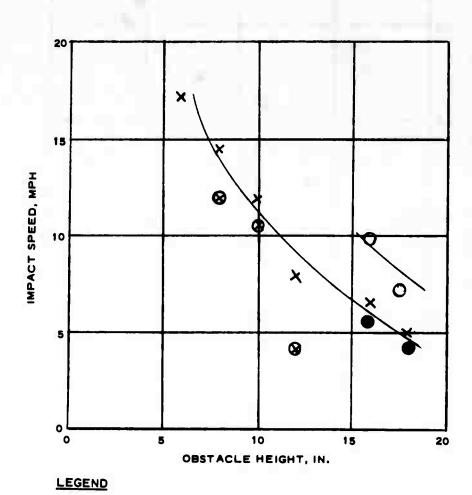
M60A1 TANK O M37 TRUCK

COMPARISON OF MEASURED AND PREDICTED SPEEDS AT WHICH PEAK VERTICAL ACCELERATION EQUALS 2.5 g's









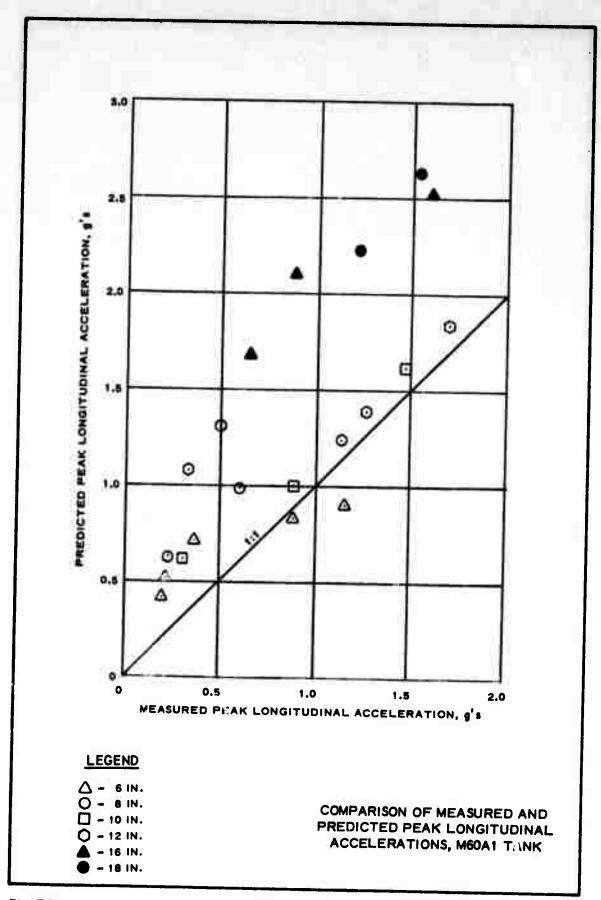
O - MEASURED, 2.0 g

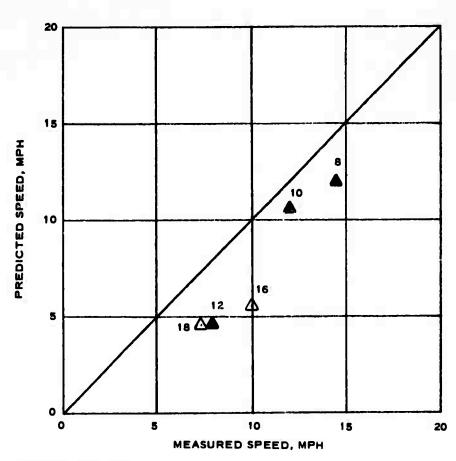
X - MEASURED, 1.0 g

S - PREDICTED, 1.0 g

- PREDICTED, 2.0 g

SPEED-OBSTACLE HEIGHT RELATIONS FOR M60A1 TANK AT 1.0-g AND 2.0-g PEAK LONGITUDINAL ACCELERATIONS





NOTE: NUMBERS NEAR PLOTTED POINTS INDICATE OBSTACLE HEIGHTS IN INCHES.

LEGEND

△ - 2.0 g

▲ - 1.0 g

COMPARISON OF MEASURED AND PREDICTED SPEEDS AT WHICH LONGITUDINAL ACCELERATION EQUALS 1.0 AND 2.0 g's

COMPUTER PROGRAM FOR SIMULATING DYNAMIC RESPONSE OF M60Al TANK, AND DICTIONARY OF PROGRAM VARIABLES

Program

TANK

```
1 SLIB, DIFFEQ
 25LIB, ALGEBR
 3$TTY, 120
 4SRPC
 5 SN DM
 6$SAV
 100
          COMMON FORCD1, FORCD2, FORCD3, FORCD4, FORCD5, FORCD6
 110
          COMMON FORCW1, FORCW2, FORCW3, FORCW4, FORCW5, FORCW6
 120
          COMMON FORCH1, FORCH2, FORCH3, FORCH4, FORCH5, FORCH6
 130
          COMMON FORCK1, FORCK2, FORCK3, FORCK4, FORCK5, FORCK6
 140
          COMMON FORTILI, FORCT1, FORCT2, FORCT3, FORCT4, FORCT5
 150
          COMMON SPDEF1, SPDEF2, SPDEF3, SPDEF4, SPDEF5, SPDEF6
 160
          COMMON DSPDF1, DSPDF2, DSPDF3, DSPDF4, DSPDF5, DSPDF6
 170
          COMMON VARI, VAR2, VAR3, VAR4, VAR5, VAR6, VAR7, VAR8, VAR9, VAR10,
 1808
            VARII, VARI2, VARI3, VARI4, VARI5, VARI6, VARI7, VARI8, VARI9
 190
          COMMON DRVI, ZETAI, HORZI, ETAI, AXLII, AXLZI, AXLZI, AXLZI, AXLZI, AXLZI
 200
          COMMON DRV2, ZETA2, HORZ2, ETA2, AXL 12, AXL 22, AXL 32, AXL 42, AXL 52, AXL 62
 210
          COMMON DDRV2, DZ ETA2, DHORZ2, DETA2, DAKL 12, DAKL 22, DAKL 32, DAKL 42,
 2208
             DAXL 52, DAXL 62
 230
          COMMON HINSTEPS, HORMOM
240
          COMMON THRESH(72), GAMMA(72), SIGMA(72), SEGDEF(72), Y(86)
          COMMON TH (4)
 250
260
          DIMENSION FORCW(6), FORCK(6), SPDEF(6), DSPDF(6)
          DIMENSION FORCH(6), FORCD(6)
270
280
          DIMENSION DISPL(10), VELCTY(10), ACCISS(10), ACCGS(10)
          DIMENSION ACCMAX(10), ACCMIN(10), SACCSQ(10), RMSACC(10)
290
300
          DIMENSION FID(12), VARID(10)
310
          EQUIVALENCE (FORCH 1, FORCH (1)), (FORCD1, FORCD(1))
320
         EQUI VALENCE (FORCWI, FORCW(1)), (FORCKI, FORCK(1))
          EQUI VALENCE (SPDEF1, SPDEF(1)), (DSPDF1, DSPDF(1))
330
         EQUI VALENCE (DRV1, DI SPL(1)), (DRV2, VELCTY(1))
340
350
         EQUIVALENCE (DDRV2, ACCISS(1))
360
         DATA I BELL/458752/
370
         DATA VARI U/5HV, DRV, 5HV, C-G, 5HH, C-G, 5HPI TCH, 5HAXLE1, 5HAXLE2,
3304
            SHAXLES, SHAXLES, SHAXLES, SHAXLE6/
         PRINT, "MURPHY'S M-60 TANK MODEL"
390
400
         PRINT
                   -----DATA INITIALIZATION-----
410C-
429
         CALL OPENF(1, "THRESH")
430
         READ( 1, ) ( THRESH( I ), I = 1, 72)
440
         CALL CLOSEF(1)
450
         CALL OPENF( 1, "GAMMA")
460
         READ(1,)(GAMMA(1), I=1,72)
470
         CALL CLOSEF(1)
430
         CALL OPENF( 1, "SIGMA")
49 0
         READ( 1, ) ( SI GMA( I ), I = 1, 72)
500
         CALL CLOSEF(1)
519
         TH(1)=13.
520
         TH(2)=10.
530
         TH(3) = 8.
```

(1 of 15 sheets)

```
TANK
        CONTINUED
540
         TH(4) = 8.
550
         DO 10 1=1.6
560
         FORCW(I) = 0.
570
         FORCK(1) = 0.
580
         SPDEF(I)=0.
590
      10 DSPDF(1)=0.
600
         DO 20 1=1.10
610
         DI SPL(1) = 0.
620
         VELCTY(I)=0.
630
         ACCISS(I) = 0.
640
         ACCGS(I)=0.
650
         RMSACC(I)=0.
660
        ACCMAX(I)=0.
670
         ACCMIN(I) = 0.
630
     20 SACCSQ(I) = 0.
690
         DO 30 I=1.86
700
     30 Y(1)=0.
710
        ZETA1 = -5.79
720
        40RZ1=0.
730
        ETA1 = - . 0089
740
        AXL 0= 13.
750
        AXL11 = -.966
760
        AXL21 = - .970
770
        AXL31 = -.942
780
        AXL41 = -.913
790
        AXL51 = -.884
300
        AXL61 = -.356
310
        TP=0.
820
        TI P= . 5
330
        T=0.
340
        DEL TAL= 33./14.
350
        NPL= 4
860
        NSTOP=0
370
        JJ=1
330
        d= . 001
890C-
                        -----DATA READ IN-----
900
        PRINT, "GIVE THE FOLLOWING INFORMATION"
910
        PRINT,
920
        PRINT, "NAME OF PROFILE INPUT FILE"
930
        READ I. FINAME
        PRINT, "TANK VELOCITY IN M.P.H."
9 40
950
        READ! XMPH
960
        "LAVATE I SMIT TUCTE INTERVAL"
970
        READ, TIP
        PRINT, "NAME OF OUTPUT DATA FILE"
980
999
        READ 1. FNAME2
1000
         PRINT, "++++END OF INPUT DATA++++"
          VEL=XMPH+17.6
1010
1020
          DEL TAT= DEL TAL/VEL
1030
         NSTEPS= DEL TAT/H
                                                                (2 of 15 sheets)
```

```
1040
           H= DEL TAT/NSTEPS
           PRINT 2, XMPH, VEL, DELTAL, DELTAT, NSTEPS, H
  1050
 1060
           WRI TE(2:11)
 1070
           CALL OPENF(1, FINAME)
 1080
           READ(1,1)FID
 1090
           PRINT,
           PRINT, "INPUT PROFILE IS:"
 1100
 1110
           PRINT 1, FID
 1120
           PRINT 3
 1130
           WRITE(2112) FID, FINAME
 1140
           WRI TE( 2; 13) XMPH, VEL, DEL TAL, DEL TAT, NSTEPS, H
 1150
           WRI TE(23 14) VARI D
 1160
           GO TO 190
      150 IF(JJ-2) 50, 40, 50
 1170
        50 READ( 1, 15) YINPUT
 1180
 1190
           CALL EDFTST(1,JJ)
 1200
           GO TO (60, 40) JJ
       40 NSTOP=NSTOP+1
 1210
 1220
       60 I=85
       70 Y(I+1)=Y(I)
 1230
 1240
           I = I - 1
1250
          IF(1)70,80,70
1260
       80 Y(1)=YINPUT
1270
           CALL DIFFER
1280
          NPL=NPL+1
1290
           T= T+ DEL TAT
1300
          DO 90 I=1,10
1310
       90 ACCGS(I)=ACCISS(I)/386.
1320
          ACCGS(4) = ACCISS(4)
1330
          DO 100 I=1,10
1340
          ACCMAX(I)=AMAX1(ACCMAX(I),ACCGS(I))
1350
          ACCMIN(I)=AMIN1(ACCMIN(I),ACCGS(I))
          SACCSQ(I)=SACCSQ(I)+ACCGS(I)+ACCGS(I)
1360
1370 100 RM SACC(I)=SORT((SACCSQ(I)+DELTAT)/T)
1380 190 IF(T-TP) 110, 120, 120
1390 120 TP= TP+ TIP
          PRINT 4, T, Y(1), (VARID(1), DISPL(1), VELCTY(1), ACCGS(1),
1400
           RM SACC(I), I=1, 10)
14108
1420 110 WRITE(2) 16) Tay(1), DISPL, VELCTY, ACCGS, RM SACC
1430
          IF(NPL-11)140, 130, 140
1440 130 WRITE(2;14) VARID
1450
          NPL=0
1460 140 IF (NSTOP-36) 150, 160, 150
1470 160 WRITE(2; 5) (VARID(I), ACCMAX(I), ACCMIN(I), I=1, 10)
         PRINT 5, (VARIDCI), ACCMAX(I), ACCMIN(I), I=1, 10)
1480
1 49 0
          IF(FNAME2-6HNOFILE) 170, 136, 170
1500 170 CALL CLOSEF(2, FNAME2, 2)
1510
         PRINT 6, FNAME2
1520 180 PRINT 7, (IBELL, I=1, 40)
1530
         CALL EXIT
```

TANK

CONTI VUED

```
CONTINUED
TANK
        1 FORMAT(12A6)
1540
       2 FORMAT(///, "VELOCITY=", F5. 2," MPH (", F6. 1," IPS )", /,
1550
           "DEL TA-L=", F5. 3, 3X, "DEL TA-T=", F6. 4,/,
15604
           "NSTEPS=", I 4, 4X, "H=", F7.6)
15704
        3 FORMAT(////2x, 4HTIME, 3X, 4HY(1), 12X, 5HDI SPL, 5X, 5HVELOC,
1580
           5X, SHACCEL, 4X, 6HRMSACC, /)
15904
        4 FORMAT(F8.4, F7.2, 2X, A5, 4G10.3, /, 9(17X, A5, 4G10.3, /))
1600
                                                      MINIMUM. /.
        5 FORMAT(37HPEAK ACCELERATIONS MAXIMUM
1610
           10(10X, A5, 2X, 2F10.3,/))
16204
        6 FORMAT( 38HCALL 2530 TO OBTAIN A LISTING OF FILE: , X, A6)
1630
1640
         7 FORMAT( 40A1)
      11 FORMAT(//, 37X, 45(1H+),/, 37X, 1H+, 43X, 1H+,/, 37X,
1650
             45H+ MURPHY'S M-60 TANK PROGRAM OUTPUT FILE
                                                                  * . / .
16608
             37X, 1d*, 43X, 1d*, /, 37X, 45(1d*), //)
16704
      12 FORMAT(16HINPUT PROFILE IS,X, 12A6,X, 11HCFILE NAME , A6, 1H],//)
1680
      13 FORMAT(9HVELOCITY=F5.2,17H MILES PER HOUR (F6.1,
1690
             19H INCHES PER SECOND) 4X, 8HDELTA-L=F5.3, 7H INCHES 4X,
17004
17104
             SHDELTA-T=F10.8.8H SECONDS//.
             35HNUMBER OF STEPS IN RKG INTEGRATION= 1 4, 4X,
17204
17304
             124STEP SIZE H=F10.8)
      14 FORMAT(//, 2X, 4HTIME3X, 4HY(1) 12X, 10(A5, 3X),//)
1740
      15 FORMAT( E20-10)
1750
      16 FORMAT(F8.4, F5.1, 2X, 6HDI SPL. 2X, 10F8.3, /, 15X, 8HVELOCI TY,
1760
           10F8.3, /, 15X, 6HACCEL.2X, 10F8.3, /, 15X, 8HRMS.ACC.10F8.3, /)
17704
1790
          END
2000
        SUBROUTINE DIFFEQ
2010
        RH=1./H
       INDEX= 0.
2020
     1 INDEX=INDEX+1
2030
2040
        VAR1=ZETA1
2050
        VAR2=ZETA2
2060
        VAR3=HORZ 1
2070
        VAR4=HORZ2
        VARS= ETA 1
2080
2090
        VAR6= ETA2
2100
           VAR19 = AXLØ
        VAR7= AXL 11
2110
        VAR8= AXL 12
2120
2130
        VAR9=AXL21
        VAR1 0= AXL 22
2140
        VARI 1= AXL 31
2150
        VAR12= AXL 32
2160
2170
        VAR13= AXL 41
2180
        VAR1 4= AXL 42
2190
        VAR1 5= AXL 51
2200
        VAR16= AXL 52
```

VAR1 7= AXL 61

VAR18=AXL62

PZETA2=ZETA2

PHORZ2=HORZ2

2210

2230

2240

(4 of 15 sheets)

```
TANK
        CONTINUED
2250
        PETA2= ETA2
2260
        PAXL 12=AXL 12
2270
        PAXL22=AXL22
2230
        PAXL 32= AXL 32
2290
        PAXL 42= AXL 42
        PAXL 52= AXL 52
2300
2310
        PAXL 62= AXL 62
        CALL ALGEBR
2320
2330
        CALL EQNS(FK12,FK14,FK16,FK18,FK110,FK112,FK114,FK116,FK113)
2340
        FK11=H+VAR2
        FK13=H+VAR4
2350
        FK15=H+VAR6
2360
2370
        FK17=H+ VAR8
2380
        FK19=H+ VAR10
2390
        FK111=H+VAR12
2400
        FK 113=H+VAR14
2410
        FK115=H+VAR16
        FK117=H+VAR18
2420
2430
        VAR1=ZETA1+FK11+.5
2440
        VAR2=ZETA2+FK12+.5
2450
        VAR3=H0RZ1+FK13+.5
2460
        VAR4#HORZ2+FK14+.5
2470
        VAR5= ETA1+FK15* . 5
2480
        VAR6= ETA2+FK 16+ . 5
2490
        VAR7=AXL11+FK17+.5
2500
        VAR8= AXL 12+FK 18+ . 5
2510
        VAR9=AXL21+FK19+.5
2520
       VAR10= AXL22+ FK 110+ . 5
2530
       VAR11=AXL31+FK111++5
2540
        VAR12=AXL 32+FK 112+ . 5
2559
       VAR13=AXL41+FK113+.5
2560
       VAR1 4= AXL 42+ FK 114+ . 5
2570
        VAR15=AXL51+FK115+.5
2580
        VAR16=AXL 52+FK116+.5
2590
       VAR17=AXL61+FK117+.5
       VAR18=AXL62+FK118+.5
2600
       CALL ALGEBR
2610
2620
       CALL EONS(FK22, FK24, FK26, FK23, FK210, FK212, FK214, FK215, FK218)
        FK21=H+VAR2
2630
2640
       FK23=H+VAR4
2650
       FK25=H+VAR6
       FK27=H+VAR3
2660
       FK29=H+VAR10
2670
2680
       FK211=H+VAR12
2690
       FK213=H+VAR14
2700
        FK215=H+VAR16
2710
        FK217=H+VAR18
       VAR1=ZETA1+.29289322+FK21+.20710678+FK11
2720
        VAR2=ZETA2+ . 29 289 322+FK22+ . 20710673+FK12
2730
        VAR3=H0RZ1+ . 29289322+FK23+ . 20710678+FK13
2740
```

(5 of 15 sheets)

```
TANK
         CONTINUED
2750
         VAR4=HORZ2+ • 29 289 322 + FK2 4+ • 20710678 + FK 14
2760
         VAR5=ETA1+ • 29 289 322+FK 25+ • 207 106 78+FK 15
         VAR6=ETA2+ . 29289322+FK26+ . 20710678+FK16
2770
2780
         VAR7=AXL 11+ • 29 289 322 + FK 27+ • 20710678 + FK 17
2790
         VAR8=AXL 12+ • 29 289 322 + FK 28+ • 20710678 + FK 18
2800
         VAR9=AXL21+.29289322+FK29+.20710678+FK19
2810
         VAR10=AXL22+ • 29 289 322+FK210+ • 20710678+FK110
2820
         VAR11=AXL31+.29289322+FK211+.20710678+FK111
         VAR12=AXL32+ • 29 259 322 + FK 212+ • 2071 0678 + FK 112
2830
         VAR13=AXL41+ . 29 289 322 + FK 213+ . 2071 06 78 + FK 113
28 40
28 50
         VAR1 4- AXL 42+ . 29 289 322+ FK 21 4+ . 20710678 + FK 11 4
2860
         VAR15=AXL51+ . 29 239 322 + FK215+ . 20710678 + FK115
2870
         VAR16=AXL 52+ . 29289 322+FK216+ . 20710678+FK116
2880
         VAR17=AXL61+ • 29289 322 + FK217+ • 20710678 + FK117
         VAR15=AXL 62+ . 29289 322+FK218+ . 20710678 + FK 118
2890
2900
         CALL ALGEBR
         CALL EQNS(FK32, FK34, FK36, FK38, FK310, FK312, FK314, FK316, FK318)
2910
2920
         FK31=H*VAR2
2930
         FK33=H + VAR4
29 40
        FK35=H+VAR6
2950
        FK37=H+VAR8
2960
        FK39=H+VAR10
2970
        FK311=H+VAR12
2980
        FK313=H+VAR14
2990
        FK315=++VAR16
3000
        FK317=H+VAR18
        VAR1=ZETA1 - . 70710678 + FK21+1.70710678 + FK31
3010
        VAR2=ZETA2- • 70710678 + FK22+ 1 • 70710678 + FK32
3020
3030
        VAR3=HORZ1- • 70710678 + FK23+ 1 • 70710678 + FK33
3040
        VAR4=HORZ2- - 70710678+FK24+1 - 70710678+FK34
3050
        VAR5=ETA1- • 70710678+FK25+1 • 70710678+FK35
3060
        VAR6=ETA2- • 70710678 + FX 26+ 1 • 70710678 + FK 36
3070
        VAR7= AXL 11- • 7071 0678 + FK27+ 1 • 7071 0678 + FK37
        VAR8=AXL12-.70710673+FK28+1.70710678+FK38
3080
3090
        VAR9=AXL21- • 70710678 + FK29+ 1 • 70710678 + FK 39
3100
        VAR10=AXL22- • 70710678 + FK210+ 1 • 70710678 + FK310
3110
        VAR1 1= AXL31- • 70710678 + FK211+1 • 70710678 + FK311
3120
        VAR12=AXL32- • 70710678 + FK212+ 1 • 70710678 + FK312
        VAR13=AXL41-.70710678+FK213+1.70710678+FK313
3130
        VAR1 4 AXL 42- • 70710678 + FK21 4+ 1 • 70710678 + FK314
3140
3150
        VAR15=AXL51-.70710678+FK215+1.70710678+FK315
        VAR16=AXL52- • 70710678 + FK216+ 1 • 70710678 + FK316
3160
        VAR17=AXL61-.70710678+FK217+1.70710678+FK317
3170
3180
        VAR18=AXL 62- • 70710678 + FK 213+ 1 • 70710673 + FK 318
3190
        CALL ALGEBR
3200
        CALL EONS (FK 42, FK 44, FK 46, FK 48, FK 410, FK 412, FK 414, FK 416, FK 418)
3210
        FK 41=H + VAR2
        FK 43= H + VAR 4
3220
3230
        FK 45=H + VAR6
3240
        FK 47=H + VAR8
```

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```
TANK
         CONTINUED
 3250
         FK 49 = H + VAR 10
 3260
         FK411=H+VAR12
 3270
         FK 413=H+VAR14
 3280
         FK 415=H+ VAR16
 3290
         FK 41 7=H+ VAR18
3300
         ZETA1=ZETA1+ • 166667 + FK 11+ • 09 763107 + FK21+ • 569 03559 + FK31
 33104
            + • 166667 + FK 41
3320
         ZETA2=ZETA2+ • 166667 * FK 12+ • 09 763107 * FK 22+ • 569 03559 * FK 32
33304
            + • 166667 + FK 42
 3340
        HORZ1=HORZ1+ • 166667+FK13+ • 09 763107+FK23+ • 569 03559+FK33
33504
            + • 166667*FK 43
        HORZ2=HORZ2+ • 166667+FK14+ • 09 763107+FK24+ • 569 03559 + FK34
3360
33704
            + • 166667 + FK 44
3380
        ETA1=ETA1+ . 166667 + FK 15+ . 09 763107 + FK25+ . 569 03559 + FK35
33904
            + • 166667 * FK 45
3400
         ETA2=ETA2+ • 166667 * FK 16+ • 09 763107 * FK 26+ • 569 03559 * FK 36
34104
            + • 166667*FK46
3420
         AXL11=AXL11+•166667#FK17+•09763107#FK27+•56903559#FK37
34304
            + • 1 66667*FK 47
3440
        AXL12=AXL12+ . 166667 + FK 13+ . 09 763107 + FK 28+ . 569 03559 + FK 38
34504
            + • 166667 + FK 48
3460
        AXL21=AXL21+ • 166667 * FK19+ • 09 763107 * FK29+ • 569 03559 * FK39
34704
            + • 166667 + FK 49
3480
        AXL22=AXL22+ • 166667 * FK110+ • 09 763107 * FK210+ • 569 03559 * FK310
34904
            + • 166667 + FK 410
        AXL31=AXL31+•160,67+FK111+•09763107*FK211+•56903559*FK311
3500
35104
            + • 166667 + FK 411
3520
        AXL32=AXL32+ • 166667 + FK112+ • 09 763107 + FK212+ • 569 03559 + FK312
35304
            + • 166667 + FK 412
3540
        AXL 41=AXL 41+ • 166667 + FK 113+ • 09 763107 + FK 213+ • 569 03559 + FK 313
35504
            + • 166667*FK413
3560
        AXL 42=AXL 42+ • 166667+F4114+ • 09 763107+FK214+ • 569 03559 +FK314
35704
            + • 166667*FK414
3580
        AXL51=AXL51+ • 166667 * FK115+ • 09 763107 * FK215+ • 569 03559 * FK315
35904
           + • 166667 + FK 415
3600
        AXL 52=AXL 52+ • 166667 * FK 116+ • 09 763107 * FK 216+ • 569 03559 * FK 316
36104
           + • 166667 + FK 416
3620
        AXL61=AXL61+.166667+FK117+.09763107+FK217+.56903559*FK317
36304
           + • 166667 + FK 417
        3640
36504
           + • 166667 + FK 418
3660
        DZ ETA2= (ZETA2-PZ ETA2) + RH
3670
        DHORZ2=(HORZ2-PHORZ2)+RH
        DETA2=(ETA2-PETA2) +RH
3680
3690
        DAXL 12= ( AXL 12- PAXL 12) + RH
3700
        DAXL 22= (AXL 22- PAXL 22) * RH
3710
        DAXL 32= ( AXL 32- PAXL 32) + RH
3720
       DAXL 42= ( AXL 42- FAXL 42) + RH
3730
        DAXL 52= (AXL 52- PAXL 52) + RH
3740
       DAXL 62= (AXL 62- PAXL 62) +RH
```

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```
TANK
       CONTINUED
3750
       DRV1 = ZETA1 + 25 * ETA1
3760
       DRV2 = ZETA2 + 25 + ETA2
3770
       DDRV2 = DZETA2 + 25 * DETA2
3780
       IF(INDEX-NSTEPS) 1, 2, 2
3790 2 RETURN
3800
       END
33 I Ø
      SUBROUTINE EQNS(FK2, FK4, FK6, FK8, FK10, FK12, FK14, FK16, FK18)
3320C
3830C----VERTICLE C-G EQUATION-----
     FK2=H+((-FORCK1-FORCK2-FORCK3-FORCK4-FORCK5-FORCK6
3840
33504
       -FORCD1-FORCD2-FORCD3-FORCD4-FORCD5-FORCD6) * · 008-386.)
3360C
3870C----HORIZONTAL C-G EQUATION----
3880 FK 4=H+( FORCH 1+FORCH 2+FORCH 3+FORCH 4+FORCH 5+FORCH 6) + . 008
389 0C
3900C----MOMENT AT C-G EQUATION----
39 10
     FK6=H*(-77. *FORCK1-44. *FORCK2-11. *FORCK3+22. *FORCK4
39204
       +55. *FORCK 5+88. *FORCK 6-77. *FORCD1-44. *FORCD2-11. *FORCD3
       +22. *FORCD4+55. *FORCD5+83. *FORCD6-HORMOM) /581700.
39 302
39 40C
3950C----BOGIE EQUATIONS (6)----
39 60
     FK8=H+(FORCK1+FORCD1+FORCT1-FORT11+FORCW1-1420.)+.2717
39 70
      FK10=H+(FORCK2+FORCD2-FORCT1+FORCT2+FORCW2-1420.)+.2717
3980
      FK12=H+(FORCK3+FORCD3-FORCT2+FORCT3+FORCW3-1420.)+.2717
3990
      FK14=d*(FORCK4+FORCD4-FORCT3+FORCT4+FORCW4-1420.) + . 2717
      FK16=H*(FORCK5+FORCD5-FORCT4+FORCT5+FORCW5-1420.) * .2717
4000
4010
      FK 18=H+(FORCK 6+FORCD6-FORCT5+FORCW6-1420.) +. 2717
4020
      RETURN
4030
      E'ND
5000
         SUBROUTINE ALGEBR
5010
          DO 95 I = 1, 6
5020
          FORCW(I) = 0.
5030
       95 FORCH(I) = 0.
5040
         VAR19= 0.
5050
         DO 20 I=1 4
5060
         IF(Y(I)-TH(I))20,20,30
5070
      30 VTH=Y(I)-TH(I)
5080
         VAR19=AMAX1(VAR19,VTH)
5090
      20 CONTINUE
5100
         DO 100 I=1,12
5110
         SEGDEF(I)=Y(I+4)-VAR7-THRESH(I)
5120
         IF(SEGDEF(I))50,60,60
5130
      50 SEGDEF(I) = 0.
5140
      60 FORCW1=FORCW1+SEGDEF(I) + GAMMA(I)
5150 100 FORCH1=FORCH1+SEGDEF(I) +SI GMA(I)
         DO 200 I=13.24
5160
5170
         SEGDEF(I)=Y(I+6)-VAR9-THRESH(I)
         IF(SEGDEF(I))150, 160, 160
5180
```

5190 150 SEGDEF(I)=0.

5200 160 FORCW2=FORCW2+SEGDEF(I) + GAMMA(I)

```
5210 200 FORCH2=FORCH2+SEGDEF(I)+SIGMA(I)
 5220
           DO 300 I=25,36
 5230
           SEGDEF(I)=Y(I+8)-VAR11-THRESH(I)
 5240
           IF(SEGDEF(I)) 250, 260, 260
 5250 250 SEGDEF(I)=0.
 5260 260 FORCW3=FORCW3+SEGDEF(I) + GAMMA(I)
 5270 300 FORCH3=FORCH3+SEGDEF(I)+SIGMA(I)
 5280
           DO 400 I = 37, 48
 529 Ø
           SEGDEF(I)=Y(I+10)-VAR13-THRESH(I)
 5300
           IF(SEGDEF(I))350,360,360
 5310 350 SEGDEF(I)=0.
 5320 360 FORCW4=FORCW4+SEGDEF(1) + GAMMA(1)
 5330 400 FORCH 4= FORCH 4+ SEGDEF(I) + SI GMA(I)
 5340
           DO 500 I = 49,60
 5350
           SEGDEF(I)=Y(I+12)-VAR15-THRESH(I)
 5360
          IF(SEGDEF(I)) 450, 460, 460
 5370 450 SEGDEF(I)=0.
 5380 460 FORCW5=FORCW5+ SEGDEF(I) + GAMMA(I)
 5390 500 FORCH5=FORCH5+SEGDEF(I)+SIGMA(I)
 5400
          DO 600 1=61,72
 5410
          SEGDEF(1)=Y(1+14)-VAR17-THRESH(1)
 5420
          IF(SEGDEF(1))550,560,560
 5430 550 SEGDEF(I)=0.
5440 560 FORCW6=FORCW6+SEGDEF(I) + GAMMA(I)
5450 600 FORCH6=FORCH6+SEGDEF(I)+SIGMA(I)
5460
          SPDEF1= VAR1+77+ + VAR5- VAR7
5470
          DSPDF1=VAR2+77. *VAR6-VAR8
5480
          SPDEF2= VAR1+44. + VAR5-VAR9
5490
          DSPDF2=VAR2+44.+VAR6-VAR10
5500
          SPDEF3= VAR1+11 + VAR5- VAR11
5510
          DSPDF3= VAR2+ 11 . * VAR6- VAR12
5520
          SPDEF 4= VAR1-22. + VAR5- VAR13
5530
          DSPDF4= VAR2- 22. + VAR6- VAR14
5540
          SPDEF5= VAR1-55. + VAR5- VAR15
5550
          DSPDF5= VAR2-55. + VAR6- VAR16
5560
          SPDEF6=VAR1-88. +VAR5-VAR17
5570
          DSPDF6= VAR2-88 . * VAR6- VAR18
5580
          DO 700 I=1.6
5590
         IF(SPDEF(I)- . 402) 710, 710, 720
5600 720 SPDEF(I)=.402
         DSPDF(I)=0.
5610
5620 710 IF(SPDEF(I)+12.) 730, 740, 740
5630 730 FORCK([)=29998 . * SPDEF([)+339972.
5640
         GO TO 700
5650 740 FORCK(I)=1667.*SPDEF(I)
5660 700 CONTINUE
5670
         DO 800 I=1.6
         IF(ABS(DSPDF(I))-1.)810,820,820
5680
5690 810 FORCD(I)=2750.+DSPDF(I)
5700
         GO TO 800
```

TANK

```
5710 820 FORCD(I)=SIGN(2750.. DSPDF(I))
 5720 800 CONTINUE
 5730
          IF(VAR7-VAR19)950,950,960
5740 950 FORT11=600.*(VAR7-VAR19)
5750
          GO TO 970
5760 960 FORT11=0.
5770 970 FORCT1 = 375. * (VAR9-VAR7)
5780
         FORCT2 = 375. * (VAR11-VAR9)
         FORCT3 = 375. + (VAR13-VAR11)
5790
         FORCT4 = 375. + (VAR15-VAR13)
58 00
         FORCT5 = 375. + (VAR17-VAR15)
5810
5820
         HORMOM= 0.
5830
         HORMOM#FORCH1#( 46.+SPDEF1+77. #VAR5)
         HORMOM=HORMOM+FORCH2+( 46++ SPDEF2+44++ VAR5)
58 40
         HORMOM=HORMOM+FORCH3+(46++SPDEF3+11++VAR5)
5850
5860
         HORMOM=HORMOM+FORCH 4+( 46++SPDEF4-22+VAR5)
5870
         HORMOM=HORMOM+FORCH5+(46++SPDEF5-55++VAR5)
        HORMOM=HORMOM+FORCH6+(46++SPDEF6-88++VAR5)
5880
5890
         RETURN
59 00
         EN D
```

THRESH

```
100 7.4,5.,3.,1.8,.6,.05,
110 .05,.6,1.8,3.,5.,7.4,
120 7.4,5.,3.,1.8,.6,.05,
130 .05,.6,1.8,3.,5.,7.4,
140 7.4,5.,3.,1.8,.6,.05,
150 .05,.6,1.8,3.,5.,7.4,
160 7.4,5.,3.,1.8,.6,.05,
170 .05,.6,1.8,3.,5.,7.4,
180 7.4,5.,3.,1.8,.6,.05,
190 .05,.6,1.8,3.,5.,7.4,
200 7.4,5.,3.,1.8,.6,.05,
210 .05,.6,1.8,3.,5.,7.4
```



GAMMA

```
100 2296, 2828, 3276, 3624, 3864, 3994, 112 3984, 3864, 3624, 3276, 2828, 2296, 120 2296, 2828, 3276, 3624, 3864, 3984, 130 3984, 3864, 3624, 3276, 2828, 2296, 140 2296, 2828, 3276, 3624, 3864, 3984, 150 3984, 3864, 3624, 3276, 2828, 2296, 160 2296, 2828, 3276, 3624, 3864, 3984, 170 3984, 3864, 3624, 3276, 2828, 2296, 182 2296, 2828, 3276, 3624, 3864, 3984, 190 3984, 3864, 3624, 3276, 2828, 2296, 200 2296, 2828, 3276, 3624, 3864, 3984, 212 3984, 3864, 3624, 3276, 2828, 2296
```

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SIGMA

```
100 7119,6150,4987,3687,2250,757,
110 -757,-2250,-3687,-4987,-6150,-7119,
120 7119,6150,4987,3687,2250,757,
130 -757,-2250,-3687,-4987,-6150,-7119,
140 7119,6150,4987,3687,2250,757,
150 -757,-2250,-3687,-4987,-6150,-7119,
160 7119,6150,4987,3687,2250,757,
170 -757,-2250,-3687,-4987,-6150,-7119,
180 7119,6150,4987,3687,2250,757,
190 -757,-2250,-3687,-4987,-6150,-7119,
200 7119,6150,4987,3687,2250,757,
210 -757,-2250,-3687,-4987,-6150,-7119
```

FIMAKE

```
1 SNDM
10C THIS PROGRAM IS USED TO MAKE OBSTACLE DAYA FILES TO BE INPUT TO 11C THE VEHICLE DYNAMICS PROGRAM.
100 DIMENSIONX(1000), FILEID(12)
110 PRINT, FILE NAME
120 READ1, FINAME
130 PRINT, FILE IDENTIFICATION"
140 READ1, FILEID
157 PRINT, "NO. OF POINTS"
170 PRINTE, N
180 READ, (X(I), I=1, N)
190 WRITE(1,1) FILEID
200 WRITE(1,3)(X(1), I=1, N)
210 PRINT, TYPE NO. OF TRAILING ZEROES
210 PRINT,
220 READ, M
230 Y=0
240 WRITE(1,3)(Y, I=1,M)
250 N= N+M
260 PRINT4, N. FINAME
270 CALLCLOSEF(1, FINAME, 2)
280 CALLEXIT
290 1FORMAT(12A6)
 300 2FORMAT (5HTYPE, 14, 12H DATA POINTS)
 310 3FORMAT(E20.10)
 320 AFOR MAT ("THERE ARE", 13, "POINTS IN FILE ", A6)
 338 END
```

Dictionary of Variables

| Variable | Description |
|------------|--|
| FORCK(I) | FORCE OF ITH SUSPENSION SPRING |
| FORCW(I) | RESULTANT VERTICAL FORCE OF SPRING SEGMENTS OF ITH BOGIE |
| FORCH(I) | RESULTANT HORIZONTAL FORCE OF SPRING SEGMENTS OF ITH BOGIE |
| FORCD(I) | FORCE OF ITH SUSPENSION DAMPER |
| SPDEF(1) | DEFLECTION OF ITH SUSPENSION SPRING |
| DSPDF(I) | VELOCITY OF ITH BOGIE DAMPER (DERIVATIVE OF SPDEF(I)) |
| FORCT(I) | TRACK TENSION FORCE BETWEEN BOGIE(I) AND BOGIE(I+1) |
| FORT11 | TRACK TENSION FORCE OF FEELER SPRING |
| VAR1-VAR19 | PAST VALUES OF ALL VARIABLES IN RUNGE KUTTA SOLUTION |
| DRV1 | VERTICAL DISPLACEMENT OF DRIVER |
| DRV2 | VERTICAL VELOCITY OF DRIVER |
| DDRV2 | VERTICAL ACCELERATION OF DRIVER |
| ZETA1 | VERTICAL DISPLACEMENT OF CG |
| ZETA2 | VERTICAL VELOCITY OF CG |
| DZETA2 | VERTICAL ACCELERATION OF CG |
| HORZ1 | HORIZONTAL DISPLACEMENT OF CG |
| HORZ2 | HORIZONTAL VELOCITY OF CG |
| DHORZ2 | HORIZONTAL ACCELERATION OF CG |
| ETA1 | PITCH DISPLACEMENT ABOUT CG |
| ETA2 | PITCH VELOCITY ABOUT CG |
| DETA2 | PITCH ACCELERATION ABOUT CG |
| AXL(I)1 | VERTICAL DISPLACEMENT OF ITH BOGIE |
| AXL(I)2 | VERTICAL VELOCITY OF ITH BOGIE |
| DAXL(I)2 | VERTICAL ACCELERATION OF ITH BOGIE |
| Н | RUNGE KUTTA TIME STEP |
| RH | RECIPROCAL OF H |
| NSTEPS | NUMBER OF STEPS IN RUNGE KUTTA SOLUTION |
| HORMOM | MOMENT OF HORIZONTAL FORCES ABOUT CG |
| THRESH(I) | THRESHOLD HEIGHTS OF 1TH SPRING SEGMENT |
| GAMMA(I) | k _v cos i |
| SIGMA(I) | khSIN i |
| SEGUEF(I) | DEFLECTION OF SEGMENT SPRING(I) |
| | (12 of 15 sheets) |

(13 of 15 sheets)

| Va | ri | ab | 1e | |
|----|----|----|----|--|
| | | | | |

Description

| Y(1) | INPUT PROFILE |
|-----------|--|
| Ti(I) | FEELER THRESHOLD HEIGHTS |
| DISPL(I) | OUTPUT DISPLACEMENTS (EQUIVALENCED TO DRV1) |
| VELCTY(I) | OUTPUT VELOCITIES (EQUIVALENCED TO DRV2) |
| ACCISS(I) | OUTPUT ACCELERATIONS IN IN./SEC2 (EQUIVALENCED TO DURV2) |
| ACCGS(I) | OUTPUT ACCELERATIONS IN g's (ACCISS(I)/386) |
| ACCMAX(I) | MAXIMUM ACCELERATION |
| ACCMIN(I) | MINIMUM ACCELERATION |
| SACCSU(I) | SUM OF THE ACCELERATIONS SQUARED |
| RMSACC(I) | RMS ACCELERATION |
| VARID(I) | BCD VARIABLE IDENTIFICATION |
| FID | BCD INPUT FILE IDENTIFICATION |
| IBELL | BCD CHARACTER TO RING TELETYPE BELL |
| TP | CONTROLS TELETYPE PRINTOUT |
| TIP | PRINTOUT TIME INTERVAL ON TELETYPE |
| T | REAL TIME |
| NPL | CONTROLS FILE PAGING |
| NSTOP | CONTROLS STOPPING OF PROGRAM |
| JJ | END OF INPUT FILE UNDICATOR |
| XMP11 | TANK SPEED IN MPH |
| VEL | TANK SPEED IN IPS |
| DELTAT | REAL TIME INCREMENT |
| DELTAL | QUANTA OF LENGTH BETWEEN INPUT PROFILE POINTS |
| FINAME | NAME OF INPUT PROFILE FILE |
| FNAME2 | NAME OF OUTPUT FILE |
| FK's | TEMPORARY DEPENDENT VARABLES IN RUNGE KUTTA |
| YINPUT | PROFILE INPUT POINT (FROM FILE) |
| INDEX | INDEX COUNTER FOR RK |

(14 of 15 sheets)

NOTE: Many of the variables stored in array's are assigned individual names as well as subscripts by use of the "EQUIVALENCE" statement. This allows for the use of either individual names or subscripts in the program whichever is more convenient.

EXAMPLE:

COMMON VNAME1, VNAME2, VNAME3
DIMENSION VNAME(3)
EQUIVALENCE (VNAME1, VNAME(1))

RESULT:

VNAME1 = VNAME(1)

VNAME2 = VNAME(2)

VNAME3 = VNAME(3)

The following variables are used in this manner:

FORCH, FORCD, FORCK, SPDEF, DSPDF, DISPL, VELCTY, AND ACCISS.

(15 of 15 sheets)

TRUCK

Program

```
ISLIB, DIFFEQ
2$LIB, ALGEBR
3$RPC
 4SN DM
 5$TTY . 120
             COMMON NSTEPS DELTAT A B MASS MASSI MASS2 CPOSI CPOS2 CHEGI CNEG2 COMMON CII C21 INTIIA FORCWI FORCW2 SPDEF1 SPDEF2 DEPDF1 DSPDF2 COMMON FORCK1 FORCK2 VAR1 VAR2 VAR3 VAR4 VAR5 VAR6 VAR7 VAR8 COMMON ZETA I ZETA 2 DZETA 2 ETA 1 ETA 2 DETA 2 NU 1 NU 2 DNU 2 COMMON MU 1 MU 2 DMU 2 COMMON THRESH (20) GAMMA (20) SEGDEF (20) Y (47) REAL NU 1 NU 2 MU 1 MU 2 MASS MASSI MASS2, INTIIA IBELL = 458752
 100
    110
    120
 130
140
150
 160
 165
 170
              PRINT 1
          1 FORMAT(18X,33(1H+),/,18X,1H*,31X,1H+,/,
18X,33H+ MURPHY S M-37 BACK-UP MODEL
 180
 1904
                 18X,1H*,3IX,1H*,/,18X,33(1H%),///)
-----DATA INITIALIZATION-
2004
21 OC
220
              A=64.8
230
              B=47.2
240
              INRTIA = 25446.
250
             MASS=8.05
260
              MASS 1=.94
270
             MASS 2= .8
280
             CPOS 1=11.8
290
              CNEG1=23.8
300
             CPOS2=10.2
310
             CNEG2=44.
320
              THRESH(1)=7.02
330
              THRESH (2)=4.5
              THRESH (3) = 2.34
340
350
              THRESH (4)=.81
360
             THRESH (5) = . 09
370
             THRESH (6) = . 09
380
             THRESH (7)=.81
390
             THRESH (8)=2.34
400
             THRESH (9) = 4.5
410
             THRESH (10)=7.02
             THRESH(11)=7.02
420
430
             THRESH(12)=4.5
             THRESH(13)=2.34
440
450
             THRESH (14)=.81
460
             THRESH (15)=.09
470
             THRESH (16)=.09
480
             THRESH(17)=.81
             THRESH (18)=2.34
490
500
             THRESH(19)=4.5
510
             THRESH (20)=7.02
520
             GAMMA(1)=411.75
530
             GAMMA (2) = 506.25
```

```
540
         GAMMA (3)=587.25
550
         GAMMA (4)=648.
560
         GAMMA (5)=668.25
570
         GAMMA (6) = 668.25
580
         GAMMA (7)=648.
590
         GAMMA (8) = 587.25
600
         GAMMA (9)=506.25
610
         GAMMA (10)=411.75
620
         GAMMA(11)=411.25
         GAMMA (12)=506.25
630
640
         GAMMA (13)=587.25
650
         GAMMA (14)=648.
660
         GAMMA (15)=668.25
670
         GAMMA (16)=668.25
         GAMMA (17)=648.
680
         GAMMA (18)=587.25
690
700
         GAMMA (19)=506.25
710
         GAMMA (20)=411,25
         ZETA 1=-5.189
720
730
         ZETA 2=0.
740
         ETA 1=.028
750
         ETA 2=0.
760
         NU 1 = -1 . 08
         NU2=0.
770
         MU1:-1.245
780
         MU 2=0.
790
800
         NSP = 0
         POINT 1=0.
810
820
         POINT 2=0.
830
         SDZ 25Q = 0.
840
         T=0.
850
         NSTOP = 0
860
         NPL=12
870
         JJ=1
         DO 10 I=1,47
880
890
     10 Y(I)=0.
                     -----DATA READ IN-----
900C ---
         PRINT, "*********TYPE IN THE DATA FOR THE FOLLOWING"
910
         PRINT, "EXTERNAL INPUT DATA"
920
930
         PRINT NAME OF PROFILE INPUT FILE READ 2. FINAME
940
950
960
       2 FORMAT(A6)
PRINT, DATA VARIABLES
970
980
990
         PRINT
          PRINT "TRUCK VELOCITY IN INCHES PER SECOND
READ VEL
PRINT "PROGRAM VARIABLES"
                                                               [REAL]
1000
1010
1020
          PRINT,
1030
```

TRUCK

CONTINUED

(2 of 9 sheets)

TRUCK CONTINUED

```
PRINT. "NUMBER OF SPACES BETWEEN PROFILE POINTS (INTEGER)
            READ NSPACE PRINT, NUMBER OF STEPS IN RKG
1050
                                                           [INTEGER]
1060
            READ NSTEPS
PRINT, EXTERNAL OUTPUT DATA
1070
1080
            PRINT, "NAME OF OUTPUT DATA FILE"
READ 2, FNAME 2
1090
1100
1110
            PRINT,
PRINT,
PRINT 3
1120
                       *******END OF DATA INPUT******
1130
1140
1150
          XMPH=VEL/17.6
3 FORMAT (////, T ", LOX, "Y(L)", 7X, "AZETA2",5X,
1160
11704
               ARMS
            WRITE(2:6)
1180
         6 FORMAT (//,37X,45(1H+),/,37X,1H+,43X,1H+,/,37X,
45H+ MURPHY'S M-37 TRUCK PROGRAM OUTPUT FILE +,/,
1190
12004
            37X, 1H*, 43X, 1H*, /, 37X, 45(1H*), ///)
WRITE(2;11)XMPH, NSTEPS
12104
1220
1230
            WRITE(2:9)
        11 FORMAT (/, 9HVELOCITY = ,F6.2,15H MILES PER HOUR,50X, 35HNUMBER OF STEPS IN RKG INTEGRATION = ,14)
1240
12504
         9 FORMAT(//,13x,9(1H-),12HDISPLACEMENT,8(1H-),3x,10(1H-),8HVELOCITY,11(1H-),3x,9(1H-),12HACCELERATION,8(1H-),2x,7HC-G RMS,/,x,4HTIME,2x,4HY(1),x,3(2x,3HC-G,4x,5HPITCH,3x,5HFR-Ax,3x,5HRE-Ax,2x),
1260
12704
12804
12904
13004
                X,5HACCEL,//)
            DELTAT=3.07/VEL
1310
            CALL OPENF(1.FINAME)
1320
          4 FORMAT (E20.10)
1330
        50 POINT 1=POINT 2
1340
        IF (JJ-2)55,40,55
55 READ(1,4)POINT2
1350
1360
            CALL EOFTST(1,JJ)
GO TO(60,40)JJ
1370
1380
1390
        40 POINT 2=POINT 1
1400
            NSTOP = NSTOP + L
            NSPACE = I
1410
        60 PSTEP=(POINT2-POINT1)/NSPACE
1420
1430
            NSP = 0
1440 100 I=46
1450
        70 Y(I+1)=Y(I)
1460
            I = I - 1
            IF(1)70,75,70
1470
1480
        75 Y(1)=POINTI
1490
            POINT 1=POINT 1+PSTEP
1500
            NSP = NSP +1
1510
            CALL DIFFEQ
1520
            SDZ 2SQ = SDZ 2SQ +DZ ETA 2*DZ ETA 2
            RMS DZ 2=SQRT ((SDZ 2SQ *DELTAT)/T)
1530
```

TRUCK CONTINUED

```
5 FORMAT(6(X,G10.4))
1540
1550
            NPL=NPL+1
            AZETA 2=DZETA 2/386.
1560
1570
            ANU2=DNU2/386.
            AMU 2=DMU 2/386.
1580
         AND Z=DMD Z/386.

RMSAZ Z=RMSDZ Z/386.

WRITE(2;7)T,POINTI,ZETAI,ETAI,MUI,NUI,ZETAZ,ETA2,MU2,NU2,

AZETAZ,DETA Z,AMUZ,ANUZ,RMSAZZ

7 FORMAT(F6.1,F5.1,13F8.3)

IF(NPL-54)120,110,120
1590
1600
16104
1620
1630
1640 110 WRITE(2;9)
            PRINT 5,T,Y(1),AZETA2,RMSAZ2
1650
1660 MPL=0
1670 128 IF(MSTOP-47)80,90,80
1660
        80 T=T+DELTAT
1680
        IF(NSP-NSPACE)100,50,100
90 CALL CLOSEF(2,FNAME2,2)
1690
1700
        PRINT 12, (IBELL, I=1,40)
12 FORMAT (40A1)
1702
1704
         PRINT 8.FNAME2
8 FORMAT(//, DETAILED OUTPUT IS IN FILE ,X,A6,/)
1710
1720
1730
            CALL EXIT
1740
            END
```

DIFFEQ

```
100
          SUBROUTINE DIFFEQ
 110
          REAL NUI, NU2, MU1, MU2, MASS, MASSI, MASS2, INRTIA
 120
          H=DELTAT/NSTEPS
 130
          RH=1./H
 140
          INDEX = 0
 150 100 INDEX=INDEX+1
 160
          VAR 1=ZETA 1
 170
          VAR 2=ZETA 2
 180
          VAR3=ETA 1
 190
          VAR 4=ETA 2
 200
          VAR5=NU1
 210
          VAR6=NU2
 220
         VAR 7=MU 1
 230
         VAR8=MU2
 240
         PZETA 2=ZETA 2
 250
         PETA 2=ETA 2
260
         PNU2=NU2
         PMU2=MU2
 270
 280
         CALL ALGEBR
290
         FK11=H+VAR2
300
         FK 12=H*(1/MASS)*(FORCK1+C11*DSPDF1+FORCK2+C21*DSPDF2-MASS*386.0)
  310
            FK 13=H +VAR 4
320
         FK | 4=H*(1/INRTIA)*(A*FORCK | +A*C| | 1*DSPDF| -B*FORCK | 2-B*C| 2| *DSPDF| 2)
  330
           FK 15=H+VAR 6
         FK16=H*(1/MASS1)*(-FORCK1-C11*DSPDF1+FORCW1-MASS1*386.0)
340
350
         FK 17=H +VAR 8
360
         FK 18=H*(1/MASS2)*(-FORCK2-C21*DSPDF2+FORCW2-MASS2*386.0)
370
         VAR I=ZETA I+FK I 1+.5
380
         VAR 2=ZETA 2+FK 12*.5
390
         VAR 3=ETA 1+FK 13*.5
400
         VAR 4= ETA 2+FK | 4*.5
410
         VAR5=NU1+FK15*.5
420
         VAR 6=NU 2+FK 16*.5
         VAR 7=MU 1+FK 17*.5
430
440
         VAR8=MU2+FK 18*.5
450
         CALL ALGEBR
460
         FK 21=H +VAR 2
         FK22=K*(I/MASS)*(FORCKI+C11*DSPDF1+FORCK2+C21*DSPDF2-MASS*386.0)
470
  48 0
           FK 23=H +VAR 4
49C
         FK 24=H*(1/INRTIA)*(A *FORCK 1+A *C11*DSPDF1-B*FORCK 2-B*C21*DSPDF2)
  500
           FK25=H*VAR6
         FK 26=H*(1/MASS1)*(-FORCK 1-C11*DSPDF1+FORCW1-MASS1*386.0)
510
520
         FK 27=H+VAR 8
530
         FK 28=H+(1/MASS 2)+(-FORCK 2-C 21+DSPDF2+FORCW2-MASS 2+386.0)
         VAR !=ZETA !+. 29289322*FK 21+. 20710678*FK !!
540
550
         VAR 2=ZETA 2+. 29289322*FK 22+. 20710678*FK 12
560
         VAR 3=ETA 1+. 29289322*FK 23+. 2071 0678*FK 13
570
         VAR 4=ETA 2+. 29289322*FK 24+. 20710678*FK 14
         VAR5=NUI+.29289322*FK25+.20710678*FK15
580
590
         VAR 6=NU2+.29289322*FK26+.20710678*FK16
```

DIFFEQ CONTINUED

```
600
         VAR 7:MU1+.29289322*FK27+.20710678*FK17
610
         VAR 8=MU 2+. 29 289 3 22 + FK 28+. 2071 06 78 + FK 18
620
         CALL ALGEBR
630
         FK 31 = H + VAR 2
         FK 32=H+(1/MASS)+(FORCK1+C11+DSPDF1+FORCK2+C21+DSPDF2-MASS+386.0)
640
  650
           FK 33=H+VAR 4
660
         FK 34=H+(1/INRTIA)+(A+FORCK:1+A+C11+DSPDF1-B+FORCK:2-B+C21+DSPDF2)
  670
           FK 35=H +VAR 6
         FK 36:H+(1/MASS 1)+(-FORCK 1-C 11+DSPDF1+FORCW1-MASS 1+386.0)
680
690
         FK 37=H*VAR 8
         FK38=H+(1/MASS2)+(-FORCK2-C21+DSPDF2+FORCW2-MASS2+386.0)
700
         VAR 1=ZETA1-.70710678*FK21+1.70710678*FK31
710
720
         VAR 2=ZETA 2-.70710678+FK 22+1.70710678+FK32
         VAR 3=ETA 1-.70710678*FK 23+1.70710678*FK 33
730
740
         VAR 4=ETA 2-.70710678*FK 24+1.70710678*FK 34
750
         VAR5=NU1-.70710678*FK25+1.70710678*FK35
         VAR 6=NU 2-.7071 0678*FK 26+1.7071 0678*FK 36
760
770
         VAR 7:MU1-.70710678*FK27+1.70710678*FK37
780
         VAR8=MU2-.70710678*FK28+1.70710678*FK38
        CALL ALGEBR
790
800
        FK 41 = H + VAR 2
        FK 42=H+(1/MASS)+(FORCK1+C11+DSPDF1+FORCK2+C21+DSPDF2-MASS+386.0)
810
  820
           FK 43=H +VAR 4
        FK 44=H+(1/INRTIA)+(A+FORCK1+A+C11+DSPDF1-B+FORCK2-B+C21+DSPDF2)
830
  840
           FK 45=H +VAR 6
        FK 46=H*(1/MASS 1)*(-FORCK 1-C 11*DSPDF1+FORCW1-MASS 1*386.0)
850
860
        FK 47=H +VAR 8
        FK 48=H + (1/MASS 2) + (-FORCK 2-C 21 + DSPDF2+FORCW2-MASS 2+386.0)
870
        ZETA 1=ZETA 1+FK 11*.166667+.09763107*FK 21+.56903559*FK 31+FK 41*.1668
880
          ZETA 2=ZETA 2+FK 12*.166667+.09763107*FK 22+.56903559*FK 32+FK 42*.16
  890
           ETA 1=ETA 1+FK 13*.166667+.09763107*FK 23+.56903559*FK 33+FK 43*.1668
  900
          ETA 2=ETA 2+FK 14*.166667+.09763107*FK 24+.56903559*FK 34+FK 44*.1668
  910
           NU1=NU1+FK15*.166667+.09763107*FK25+.56903559*FK35+FK45*.166667
  920
           NU2=NU2+FK16+.166667+.09763107+FK26+.56903559+FK36+FK46+.166667
  930
          MU1=MU1+FK17+.166667+.09763107+FK27+.56903559+FK37+FK47+.166667
  940
          MU2:MU2+FK18*.166667+.09763107+FK28+.56903559*FK38+FK48*.166667
  950
          DZETA 2= (ZETA 2-PZET.4 2) +RH
  960
970
        DETA 2= (ETA 2-PETA 2) *RK
980
        DNU2=(NU2-PNU2) *RH
        DMU2=(MU2-PMU2)+RH
990
1000
         IF (INDEX-NSTEPS)100,200,200
1010 200 RETURN
1020
         END
```

ALGEBR

```
SUBROUTINE ALGEBR
100
110
         REAL NUI, NU2, MU1, MU2, MASS, MASSI, MASS2, INTIIA
         DO 100 I=1,10
SEGDEF(I)=Y(I)-VAR5-THRESH(I)
120
130
         IF(SEGDEF(I))50,100,100
140
150
     50 SEGDEF(I)=0.
160 100 CONTINUE
170
         DO 200 I=11,20
         SEGDEF (I)=Y(I+26)-VAR7-THRESH(I)
180
190
         IF(SEGDEF(I))150,200,200
200 150 SEGDEF(I)=0.
210 200 CONTINUE
220
         FORCWI = 0.
         FORCW2=0.
230
         DO 300 I=1,10
FORCW1=FORCW1+GAMMA(I)*SEGDEF(I)
240
250
260 300 CONTINUE
         DO 400 I=11,20
FORCW2=FORCW2+GAMMA(I)*SEGDEF(I)
270
280
290 400 CONTINUE
300
         SPDEF1=VAR5-VAR1-A+SIN(VAR3)
         SPDEF 2=VAR 7-VAR 1+B+SIN (VAR 3)
310
         DSPDF1=VAR6-VAR2-A+VAR4+COS(VAR3)
320
330
         DSPDF2=VAR8-VAR2+B+VAR4+COS(VAR3)
         FORCK 1=19.54*(SPDEF1*+3)-192.42*SPDEF1*SPDEF1+913.55*SPDEF1
340
350
         FORCK 2=1.39 * (SPDEF2 * + 3) - 1.24 * SPDEF2 * SPDEF2 + 307.72 * SPDEF2
360
         IF(DSPDF1)600,500,500
370 500 C11=CPOS1
         GO TO 700
380
390 600 C11=CNEG1
400 700 IF(DSPDF2)800.900.900
410 900 C21=CP0S2
420
        RETURN
430 800 C21=CNEG2
440
        RETURN
450
        END
```

Dictionary of Variables

| Variable | |
|----------|---|
| NSTEPS | NUMBER OF STEPS IN RUNGE KUTTA SOLUTION |
| D ELTAT | REAL TIME INCREMENT |
| A | DISTANCE FROM FRONT AXLE TO CG |
| В | DISTANCE FROM REAR AXLE TO CG |
| MASS | MASS OF VEHICLE |
| MASSI | MASS OF FRONT AXLE |
| MASS2 | MASS OF REAR AXLE |
| CPOSI | DAMPING COEFFICIENT FOR POSITIVE MOTION OF FRONT AXLE |
| C POS2 | DAMPING COEFFICIENT FOR POSITIVE MOTION OF REAR AXLE |
| CNEGI | DAMPING COEFFICIENT FOR NEGATIVE MOTION OF FRONT AXLE |
| C NEG2 | DAMPING COEFFICIENT FOR NEGATIVE MCFION OF REAR AXLE |
| N SPACE | NO. OF INTERPOLATION POINTS BETWEEN PROFILE POINTS |
| VEL | VELOCITY |
| ZETAI | VERTICAL DISPLACEMENT OF CG |
| ZETA2 | VERTICAL VELOCITY OF CG |
| DZETA2 | VERTICAL ACCELERATION OF CG |
| AZETA2 | VERTICAL ACCELERATION IN G'S OF CG |
| ETAI | PITCH DISPLACEMENT ABOUT CG |
| ETA2 | PITCH VELOCITY ABOUT CG |
| DETA2 | PITCH ACCELERATION ABOUT CG |
| NUI | VERTICAL DISPLACEMENT OF FRONT AXLE |
| N U2 | VERTICAL VELOCITY OF FRONT AXLE |
| DNU2 | VERTICAL ACCELERATION OF FRONT AXLE |
| A NU2 | VERTICAL ACCELERATION IN G'S OF FRONT AXLE |
| MUI | VERTICAL DISPLACEMENT OF REAR AXLE |
| MU2 | VERTICAL VELOCITY OF REAR AXLE |
| DMU2 | VERTICAL ACCELERATION OF REAR AXLE |
| A MU2 | VERTICAL ACCELERATION IN G'S OF REAR AXLE |
| Ã | PROFILE SHIFT REGISTER |
| | |

(8 of 9 sheets)

| Variable | Description |
|----------|--|
| POINTI | PAST PROFILE POINT |
| POINT2 | PRESENT PROFILE POINT |
| PSTEP | PROFILE INTERPOLATION INCREMENT =[(POINT2 - POINTI)/NSPACE] |
| SD72SQ | SUM OF THE SQUARES OF DZETA2 |
| RMSDZ2 | RMS OF DZETA2 [RMS (in./sec ²)] RMSAZ2 RMS (g) OF CG |
| FINAME | NAME OF PROFILE FILE |
| FNAME | NAME OF OUTPUT FILE |
| NSP | PRESENT STEP IN PROFILE INTERPOLATION |
| JJ | END OF PROFILE FILE INDICATOR |
| NSTOP | STEP COUNTER FOR T AFTER LAST POINT IS READ FROM PROFILE FILE* |
| I | NO SPECIAL SIGNIFICANCE USED FOR ALL "DDloop" COUNTS |

^{*} PROGRAM STOPS AFTER LAST PROFILE POINT IS READ AND SHIFTED THROUGH Y.